



## **Autoevaluation test. Chapter 1.**

### **1. Define the Phases of structural design**

- Specification of function and design criteria
- Determination of applied load
- Calculation of internal element loads
- Determination of allowable element strength
- Experimental test

### **2. Explain the differences between surface and body loads.**

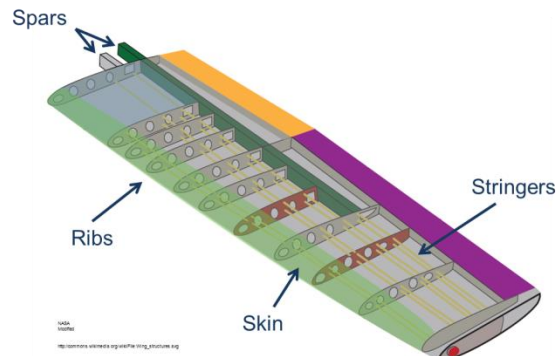
Surface forces are those that act on a surface. Body loads are those that act throughout a volume by a long-range interaction with matter or charges at a distance

### **3. Explain the necessity of introducing a safety factor in structural design.**

- Limitations of calculus model
- Variation in physical properties of materials
- Variation in fabrication standard
- Emergency conditions
- Variation of loads (gust random

**4. Describe the elements of a cantilever wings.**

The principal structural parts of the wing are spars, ribs, and stringers



Principal structural elements of the wing

**5. Explain the main functions of the ribs in a wing.**

Ribs give the wing section its shape and support in-plane loads. They transmit the load applied to the wing, from the skin to the spars. Also, ribs reduce the buckling length of the stringers and spars.

**6. List the loads applied on the fuselage**

- Wing reactions
- Landing gear reactions
- Internal pressure

## 7. List the Fuselages types

### Non stressed-skin structures

- Truss fuselage
- Geodetic fuselage

### Stressed-skin Structures

- Corrugated (3)
- Monocoque fuselage
- Semi-Monocoque fuselage

## 8. Describe the advantages and disadvantage of the Geodetic fuselage design

### Advantages:

Lightweight structure

High strength

Damage tolerance

### Disadvantages:

Complexity

## 9. Explain the differences in stabilizer design between civil and military aircraft (fighters).

In fighter design, usually no spar + ribs construction is used

## 10. Explain the differences between conventional a T-tail design.

The span of the t-tail fin is aprox.  $1/3$  than conventional tail

The vertical stabilizer must be made considerably stronger and stiffer to support the forces generated by the tailplane