



Universidad
Carlos III de Madrid
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Aerospace Structures

Chapter 1. Structural description of the aircraft



CHAPTER 1. Structural description of the aircraft

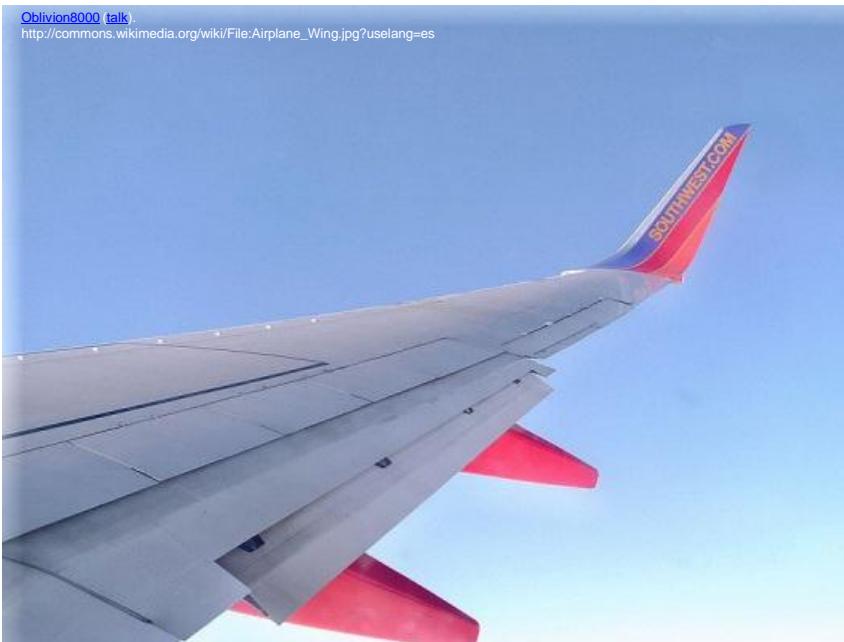
Structural components of aircraft

- Introduction
- Wing structure
- Fuselage structure
- Stabilizers structure
- References



Wing structure

■ Wing function



The function of the wing is to produce lift

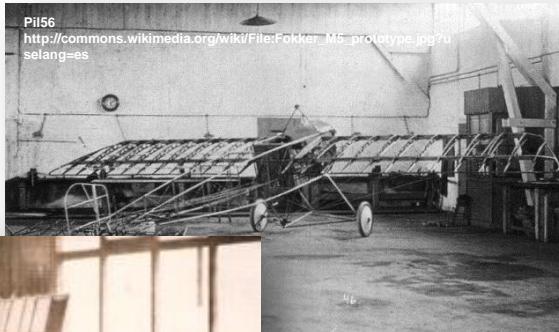
Main loads applied on the wing:

- Aerodynamic loads
- Weight

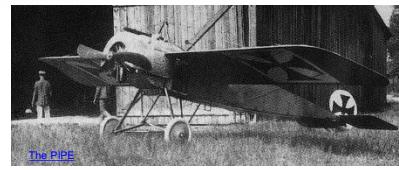
A wing works as a beam able to support the internal forces (Shear, bending and torsional moments)

■ Structures types

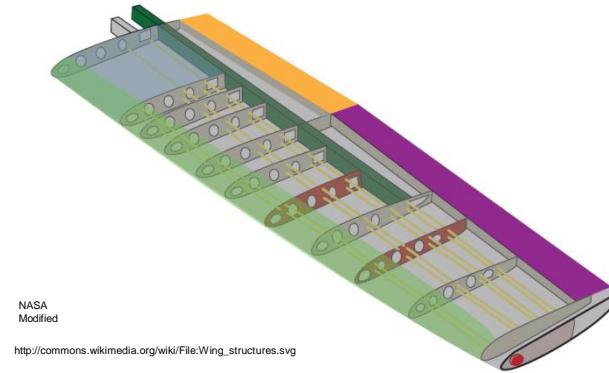
Truss-type construction



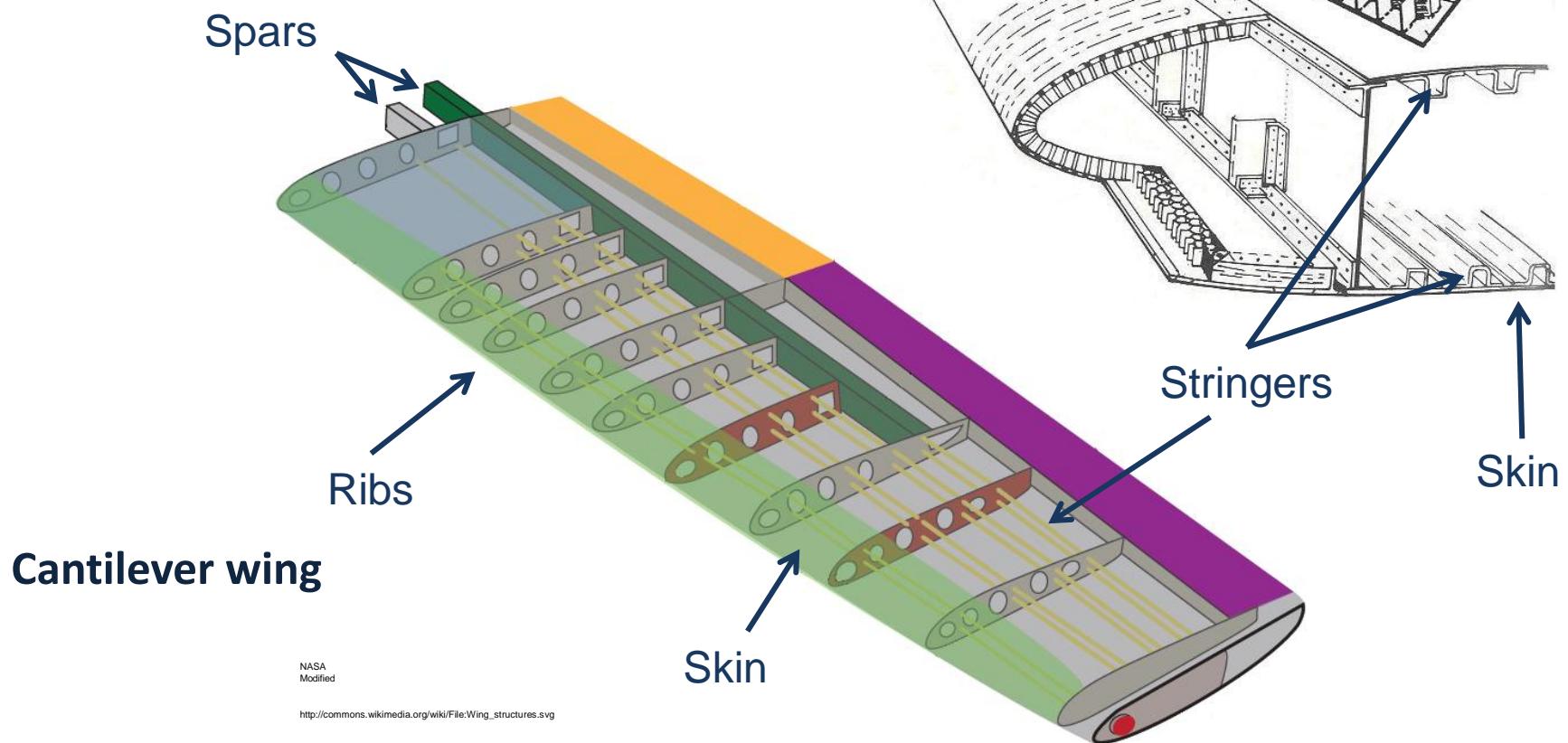
Early wing structures



Stressed-skin construction

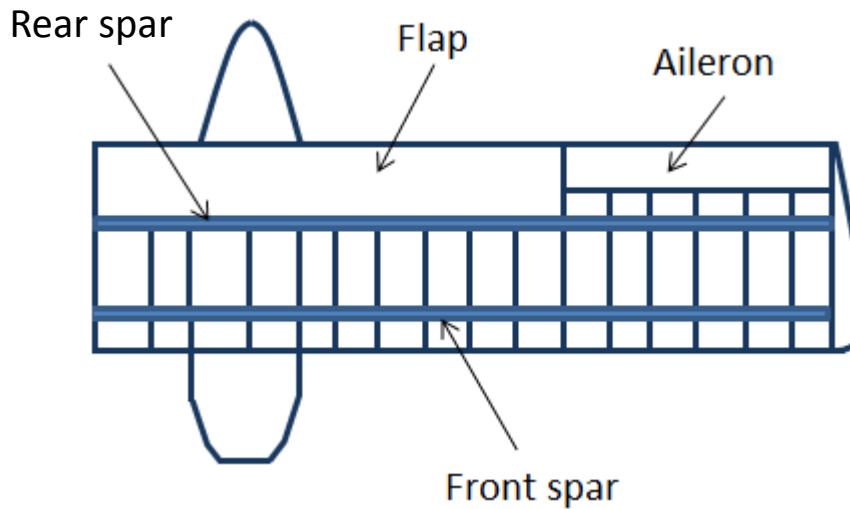


■ Wing elements



Principal structural elements of the wing

■ Spars



Torsion box structure: 2-3 spar

Front spar: 15-30% chord

Rear spar: 65-75% chord

Semicantilever wings

Some aircraft have external struts for wing bracing



bus227
<http://commons.wikimedia.org/wiki/File:FlyingWires.JPG>



CambridgeBayWeather



NASA/The Boeing Company
http://commons.wikimedia.org/wiki/File:Boeing_SUGAR_Volt_concept_aircraft_2010.jpg?uselang=es



Credit: NASA Langley/Sean Smith
http://www.nasa.gov/centers/langley/multimedia/iotw-tdt-wing_pr.htm

Boeing Aerodynamic Efficiency
Improvement Joined Wing

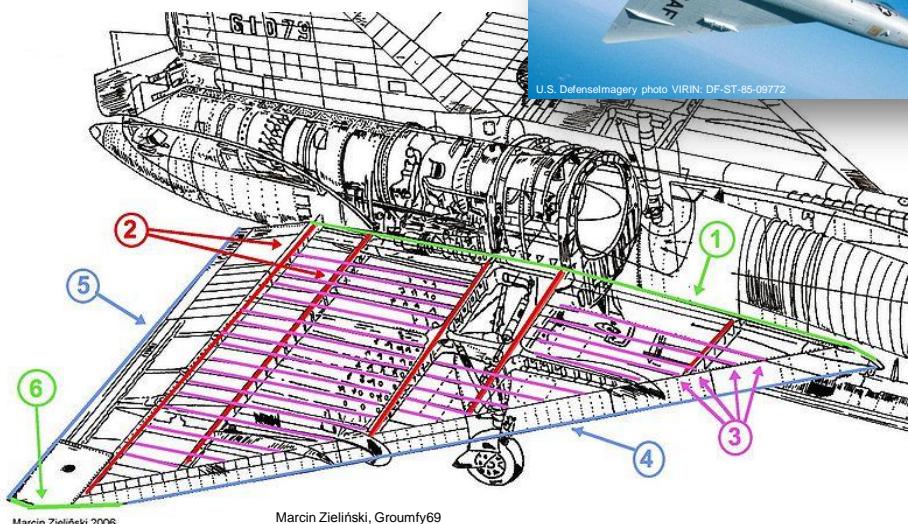


Boeing SUGAR Volt concept

Wing structure

■ Spars

Convair F-106A Delta Dart

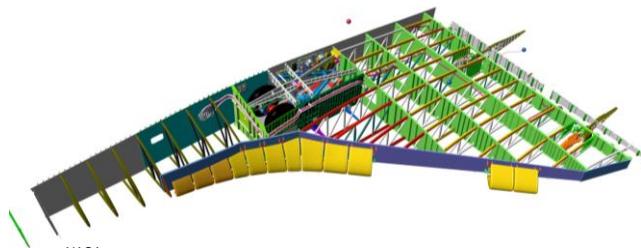


- (1) Wing root
- (2) Spars
- (3) Ribs
- (4) Leading edge
- (5) Trailing edge
- (6) Wing tip

<http://commons.wikimedia.org/wiki/File:Delta-Dart-DF-ST-85-09772.JPG?uselang=es>



Torsion box structure: multispar
Parallel spars



NASA
http://commons.wikimedia.org/wiki/File:Shuttle_Left_Wing_Cutaway_Diagram.jpg?uselang=es
http://www.nasa.gov/columbia/home/COL_wing_diagrams.html



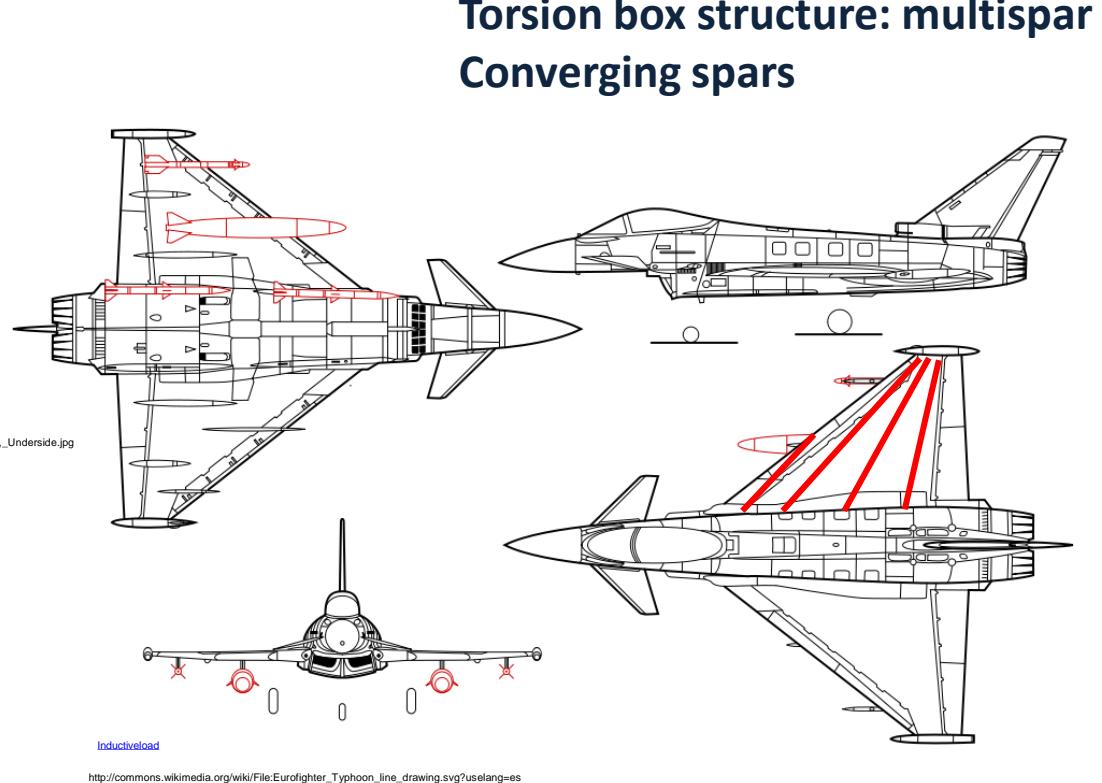
NASA Photo ID: S81-30746
http://commons.wikimedia.org/wiki/File:Columbia_landing_on_Rogers_dry_lake.triddle.jpg

■ Spars



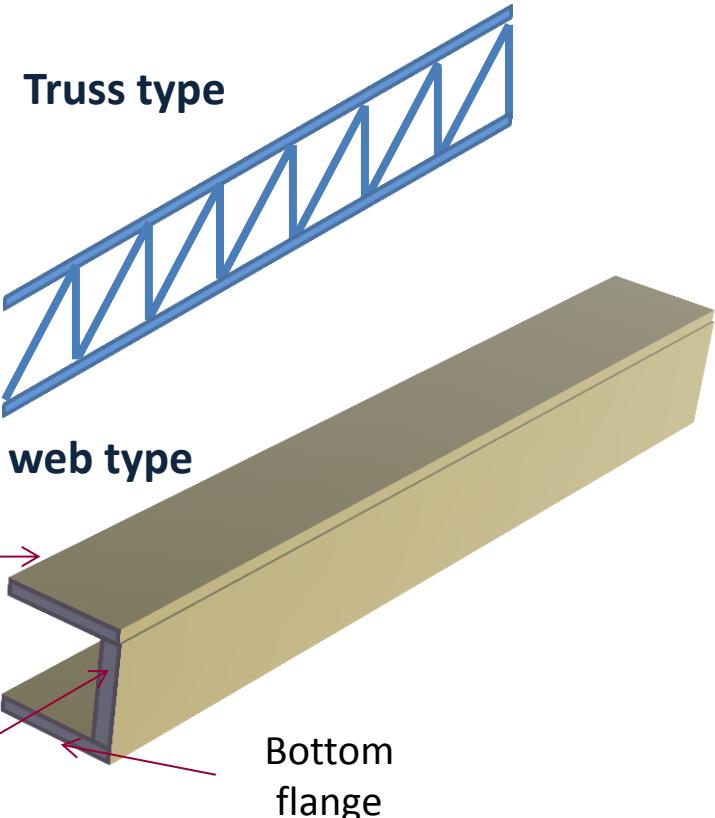
tMH
http://commons.wikimedia.org/wiki/Category:Eurofighter_Typhoon?uselang=es#mediaviewer/File:Eurofighter_Typhoon,_Underside.jpg

Eurofighter

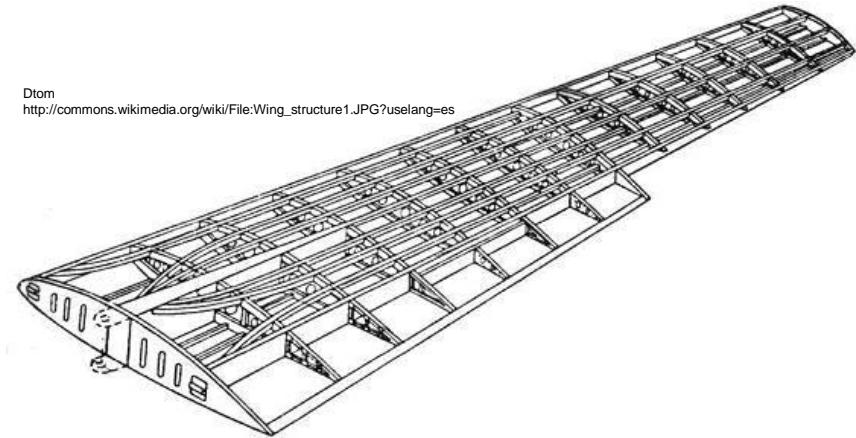


■ Spars

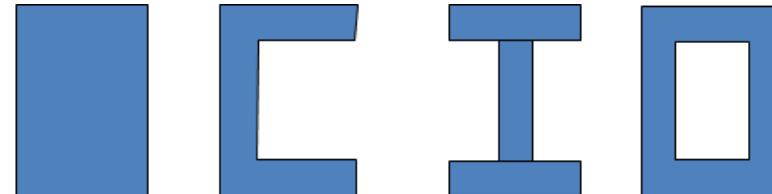
Types of spar construction



Dtom
http://commons.wikimedia.org/wiki/File:Wing_structure1.JPG?uselang=es



Cross-section geometries



Rectangular

Channel

I or double-T

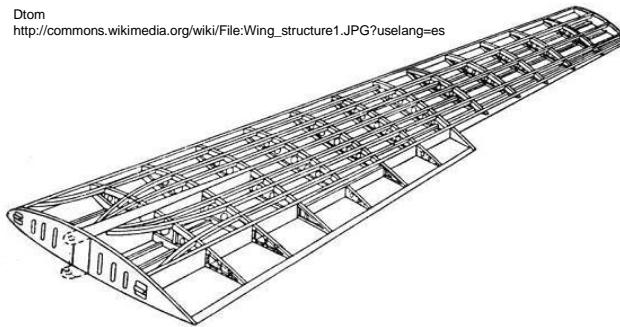
Box

Wing structure

■ Ribs

Dtom

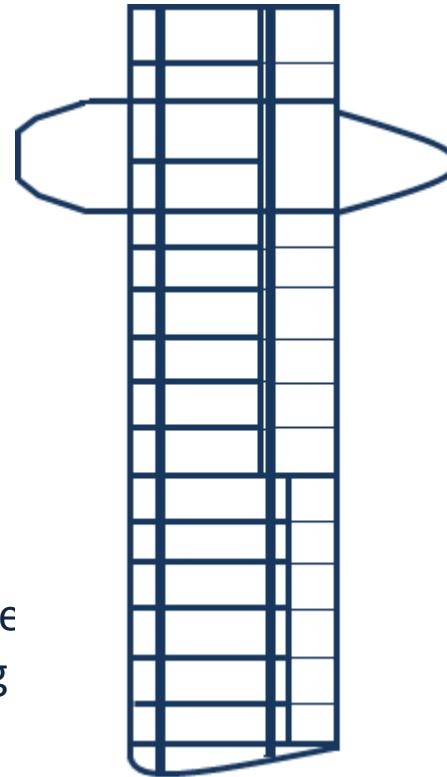
http://commons.wikimedia.org/wiki/File:Wing_structure1.JPG?uselang=es



The wing ribs determine the shape and thickness of the wing

Ribs types

- Shear web ribs
- Truss ribs



The rib spacing is selected after a optimization process (skin+stringer+ribs)

Rib are to be located at each aileron and flap hinge.

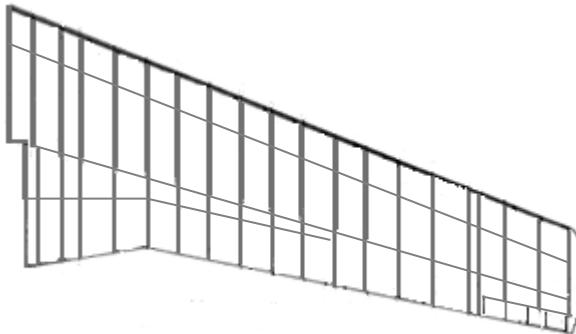
Wing structure

■ Ribs arrangement



http://commons.wikimedia.org/wiki/File:Aerospatiale_SE-210_Caravelle_10B3_Super_B,_Sterling_Airways_AN0018394.jpg?uselang=es

Aerospatiale Caravelle

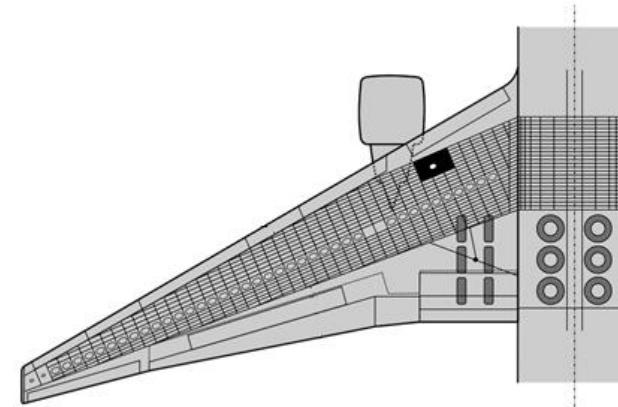


Wing ribs parallel to the flight path



[http://commons.wikimedia.org/wiki/File:Boeing_767#mediaviewer/File:LOT_Polish_Airlines,_Boeing_767-25D_ER,,Mississauga_\(295649204\).jpg](http://commons.wikimedia.org/wiki/File:Boeing_767#mediaviewer/File:LOT_Polish_Airlines,_Boeing_767-25D_ER,,Mississauga_(295649204).jpg)

Boeing 767

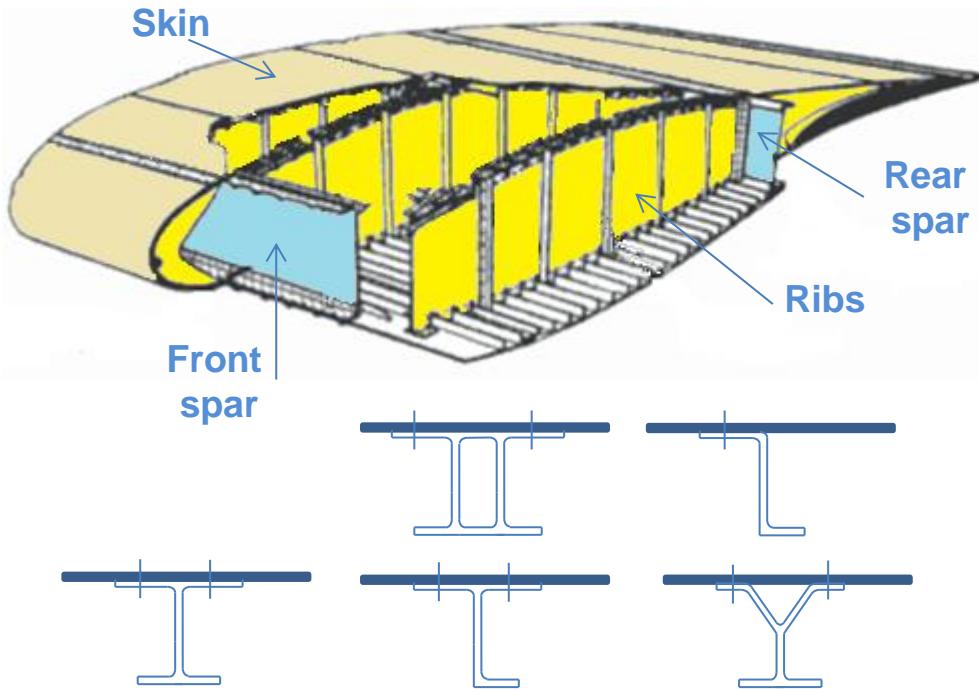


Tosaka
[http://commons.wikimedia.org/wiki/File:Jetliner%27s_wing_structure_\(B-777\).PNG](http://commons.wikimedia.org/wiki/File:Jetliner%27s_wing_structure_(B-777).PNG)

Rib perpendicular to the spars

■ Skin

50-70% of the structural weight of the wing



Typical wing skin-stringer panels

It transmits aerodynamic forces to the longitudinal and transverse members

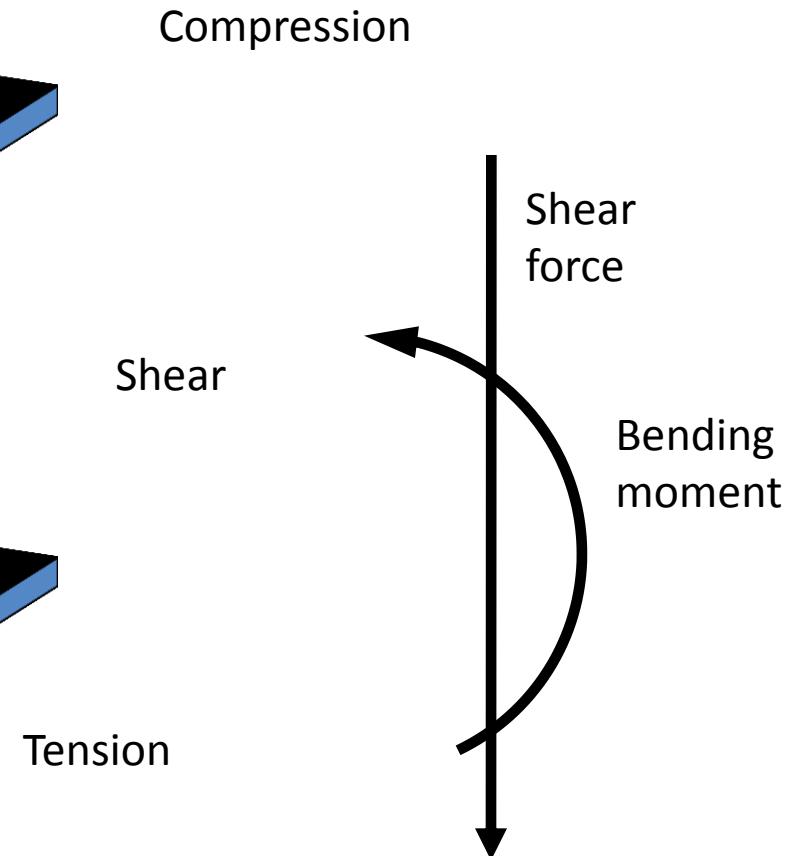
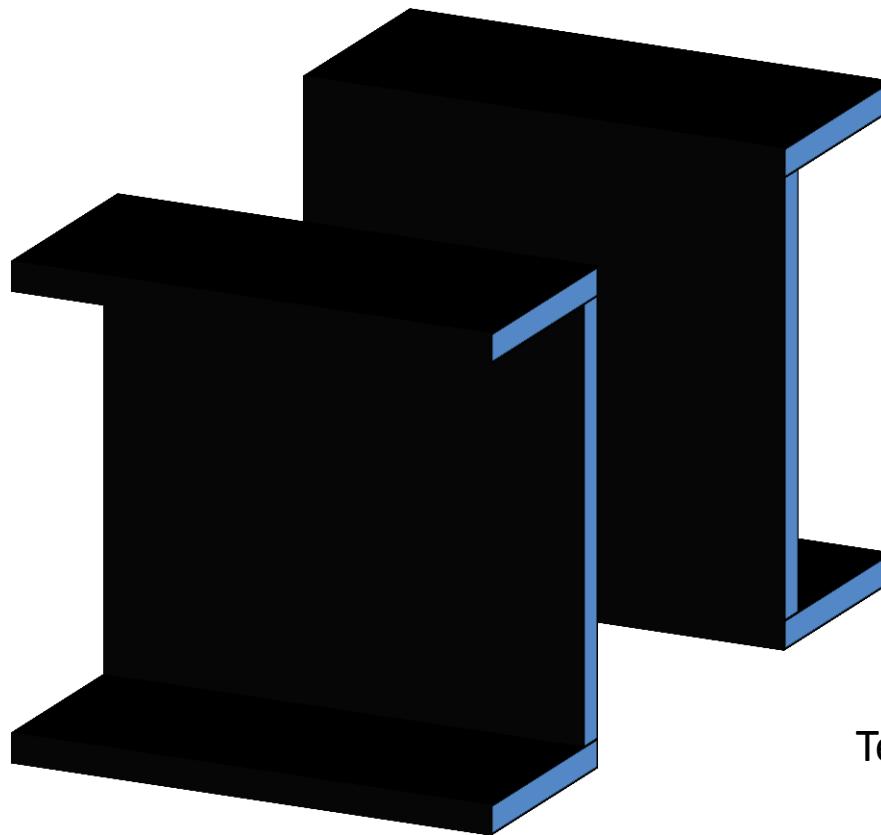
In some designs the spar caps resist the bending stresses, in other skin is the bending-load resistant element

Integrally stiffened panels

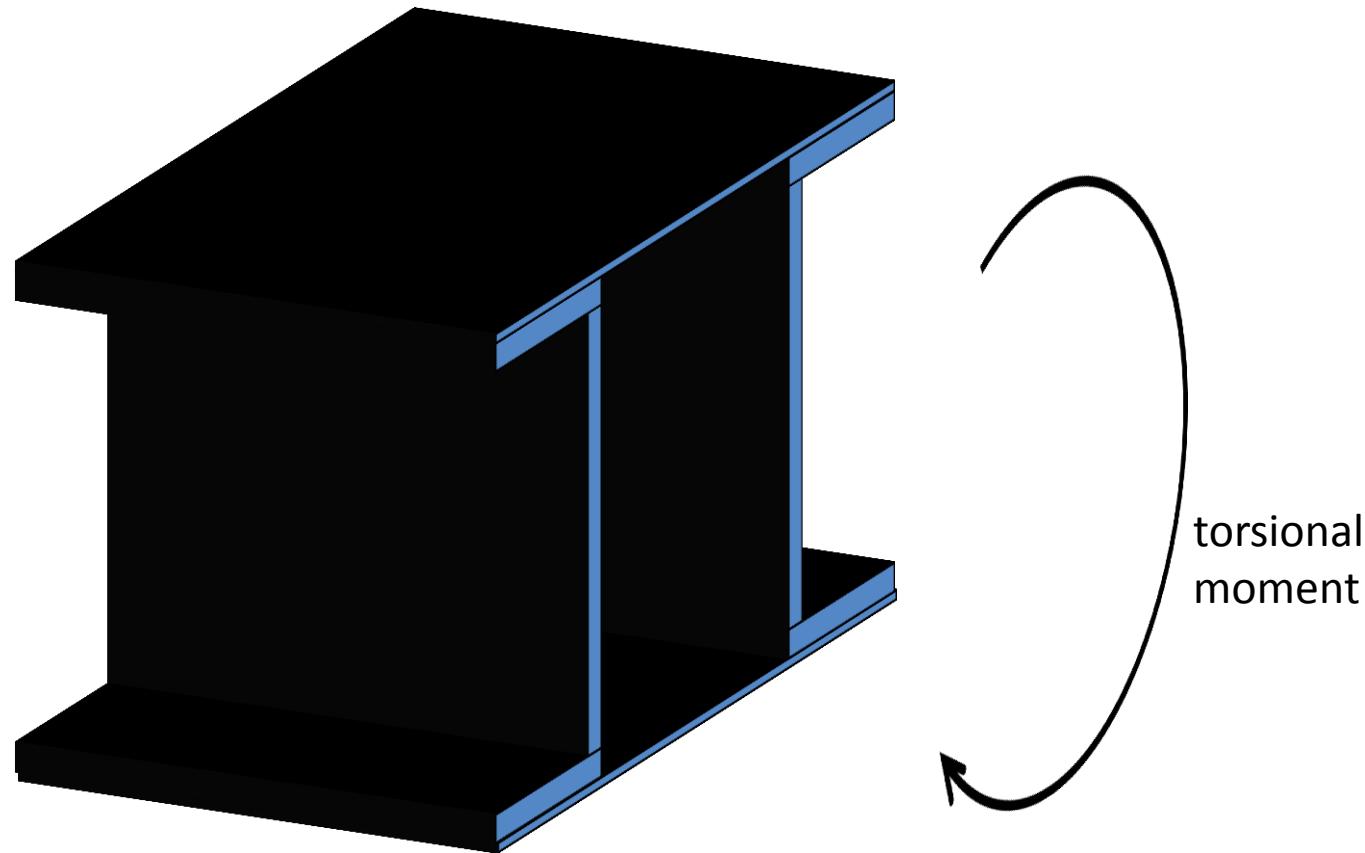


Wing skins are machined from a thick plate to obtain the skin-stringer geometry

■ Wing box structure



■ Wing box structure





Fuselage structure

Fuselage structure

■ Fuselage functions



The fuselage function is to carry the payload and support many the aircraft system. They connect the other structural element (wing and empennage)



Loads on the fuselage:

- Wing reactions
- Landing gear reactions
- Internal pressure

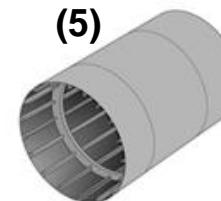
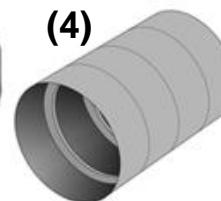
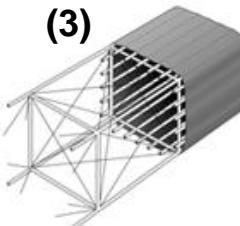
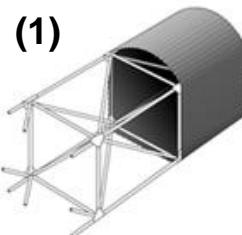
■ Fuselage types

Non stressed-skin structures

- Truss fuselage (1)
- Geodetic fuselage (2)

Stressed-skin Structures

- Corrugated (3)
- Monocoque fuselage (4)
- Semi-Monocoque fuselage (5)



Author: Tosaka
[http://commons.wikimedia.org/wiki/File:Airframe_\(4_types\).PNG?uselang=es](http://commons.wikimedia.org/wiki/File:Airframe_(4_types).PNG?uselang=es)

Blériot XI

(1)



Kogo

http://commons.wikimedia.org/wiki/B%CC%A9ri%C3%A9ot_XI?uselang=es#mediaviewer/File:Bleriot_XI_Thulin_2.jpg

Vickers Wellington

(2)



British Government.

http://commons.wikimedia.org/wiki/File:Vickers_Wellington_Mk2.jpg

Ford Trimotor

(3)



http://commons.wikimedia.org/wiki/File:Ford_Trimotor_EAA.jpg?uselang=es

Rutan VariEze

(4)



Stephen Kearney

http://commons.wikimedia.org/wiki/Category:Rutan_VariEze#mediaviewer/File:VariEze_in_flight.jpg

Airbus A350

(5)



Don-vip
http://commons.wikimedia.org/wiki/Airbus_A350#mediaviewer/File:A350_First_Flight_-,_Low_pass_03.jpg

Fuselage structure

■ Fuselage types

Truss fuselage



Philip Capon

http://commons.wikimedia.org/wiki/File:Sopwith_Camel_taking_off,_Masterton,_New_Zealand,_April_2009.jpg?uselang=es

Sopwith Camel, 1917

Most of the early aircraft used this kind of construction



[http://commons.wikimedia.org/wiki/File:Sopwith_Camel_\(repro\)_structure.JPG?uselang=es](http://commons.wikimedia.org/wiki/File:Sopwith_Camel_(repro)_structure.JPG?uselang=es)

TSRL

The diagonal elements can be bars or wires

□ Fuselage types

Truss fuselage

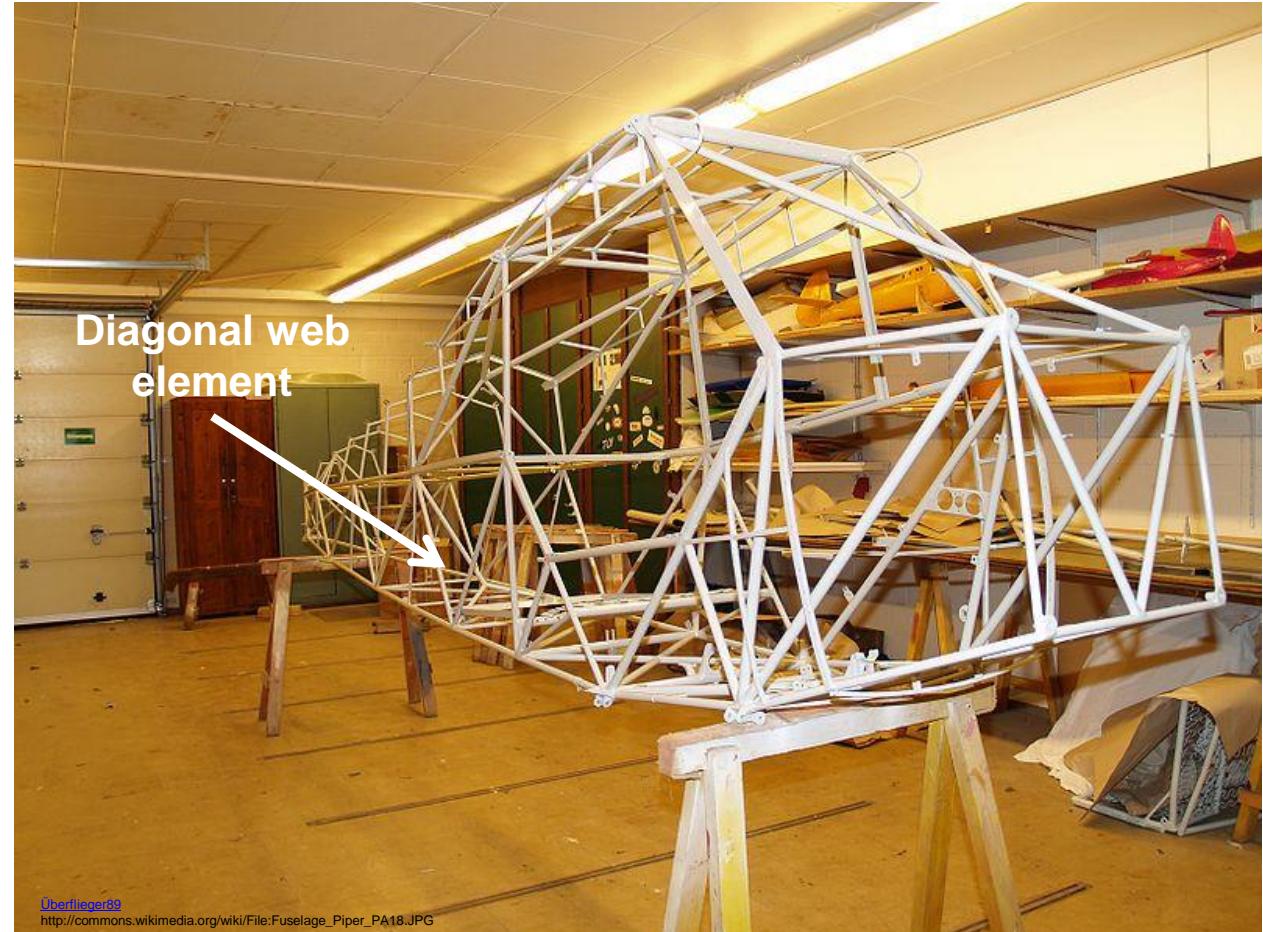


http://commons.wikimedia.org/wiki/Category:Super_18_Model_S18#mediaviewer/File:Super18-Super18-180.jpg

Photo: FlugKerl2

Piper PA-18

This type of structures are still used in small aircraft



□ Fuselage types

Truss fuselage

Non structural skin



http://commons.wikimedia.org/wiki/Piper_PA-18#mediaviewer/File:L18FuselageCUT.jpg

Überflieger89



http://commons.wikimedia.org/wiki/Piper_PA-18#mediaviewer/File:L18FuselageCT.jpg

Überflieger89

Fuselage structure

□ Fuselage types

Geodetic fuselage



Vickers Wellington
1936



Cobaltor
[http://commons.wikimedia.org/wiki/File:Vickers_Wellington_Mark_X,_HE239_%27NA-Y%27,_of_No._428_Squadron,_RCAF_\(April_1943\).png?uselang=es](http://commons.wikimedia.org/wiki/File:Vickers_Wellington_Mark_X,_HE239_%27NA-Y%27,_of_No._428_Squadron,_RCAF_(April_1943).png?uselang=es)



Authors: Enrique Barbero Pozuelo, José Fernández Sáez, Carlos Santiuste Romero

Advantages:

- Lightweight structure
- High strength
- Damage tolerance

Disadvantages:

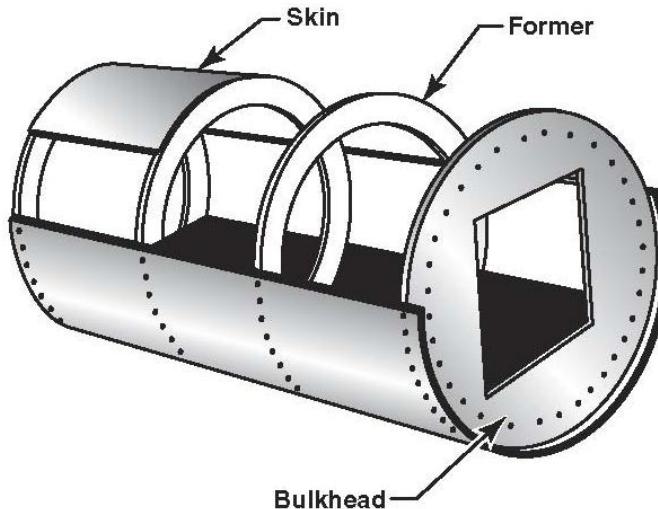
- Complexity

In this type of construction multiple flat strip stringers are wound about the formers in opposite spiral directions

Fuselage structure

□ Fuselage types

Eas4200c.f08.nine.o
http://commons.wikimedia.org/wiki/File:Chapter_1_img_26.jpg?uselang=es



The skin supports the primary stresses



Deperdussin monoplane racer
1912

First monocoque design

Designed by Louis Béchereau

Monocoque fuselage

Virtually no internal framework

Only transversal elements (former) + skin

Stoddard-Hamilton Glasair III



http://commons.wikimedia.org/wiki/Category:Stoddard-Hamilton_Glasair_III#mediaviewer/File:GlasairIII-Landing.jpg

Used today in small aircrafts with complete composite structure

Fuselage structure

□ Fuselage types

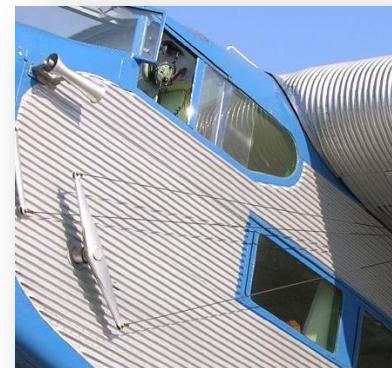


http://commons.wikimedia.org/wiki/File:Ju-Air_Junkers_Ju-52-3m_HB-HOS_Hahnweide_2011_03.jpg?uselang=es

Junkers Ju-52/3m
1932



http://commons.wikimedia.org/wiki/File:Ju52_civil.jpg?uselang=es



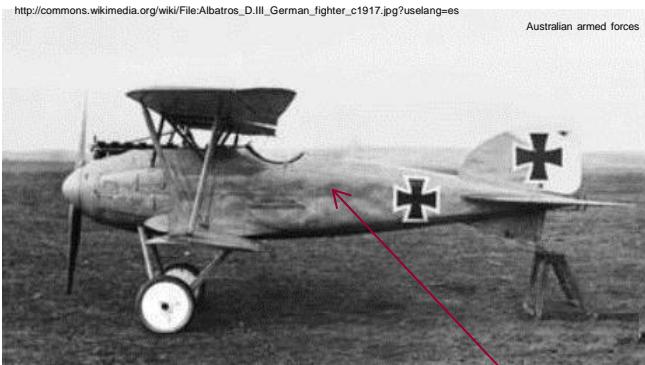
Ford Tri-Motor
1926



[http://commons.wikimedia.org/wiki/File:Ford_Tri-Motor_\(EAA\)_%28340406538%29.jpg?uselang=es](http://commons.wikimedia.org/wiki/File:Ford_Tri-Motor_(EAA)_%28340406538%29.jpg?uselang=es)

It had a structure with
metallic corrugated skin

□ Fuselage types



Albatros D.III
1916

Streamlined fuselage

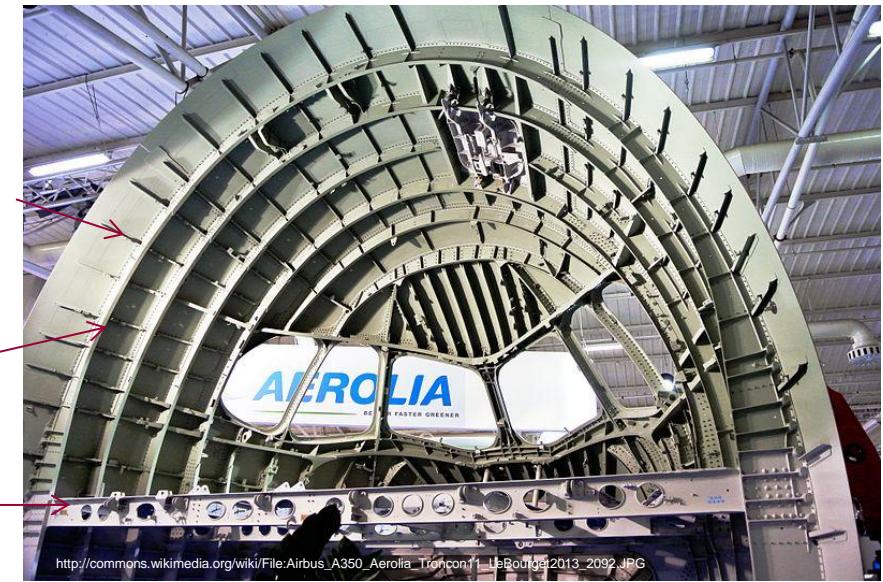
Longitudinal elements

- Longerons
- Stringers

Transverse element

- Frames
- Bulkheads

Semi-Monocoque fuselage

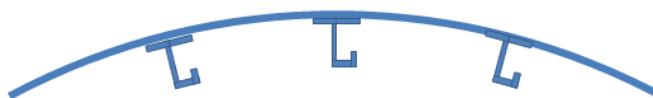


http://commons.wikimedia.org/wiki/File:Airbus_A350_Aerolia_Troncon11_LeBouget2013_2092.JPG

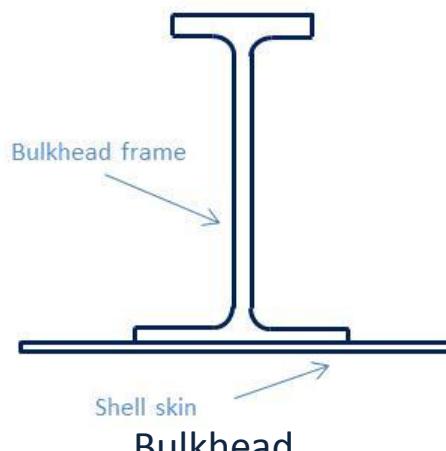
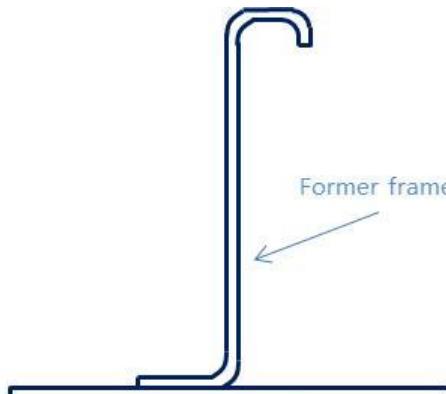
□ Skin and stringer

The skin carries the shear stresses produced by the internal forces (torque and shear force)

**Typical fuselage
skin-stringer panels**



□ Frames



Author: Kolossos
<http://commons.wikimedia.org/wiki/File:Fuselage-747.jpg?uselang=es>



Distance between frames are around 500 mm

Frame thickness are 75-150 mm

Frames function

- Maintain the shape of the fuselage
- Increase the buckling strength of the stringer



Bibi95
[http://commons.wikimedia.org/wiki/File:Exposition_-_Les_100_ans_de_l%C2%A9rospace_-_Paris_-_4_Octobre_2008_\(2913799759\).jpg](http://commons.wikimedia.org/wiki/File:Exposition_-_Les_100_ans_de_l%C2%A9rospace_-_Paris_-_4_Octobre_2008_(2913799759).jpg)

□ Frames

Bibi95

http://upload.wikimedia.org/wikipedia/commons/1/19/Exposition_Les_100_ans_de_l_aerospatiale._-Paris._-4_Octobre_2008_%282913796295%29.jpg



Front dome

Rear dome



Frames →

Types of frames

- Former frame
- Bulkhead
- domes

NASA
http://commons.wikimedia.org/wiki/File:Shuttle_Carrier_Aircraft_interior_bulkhead.jpg

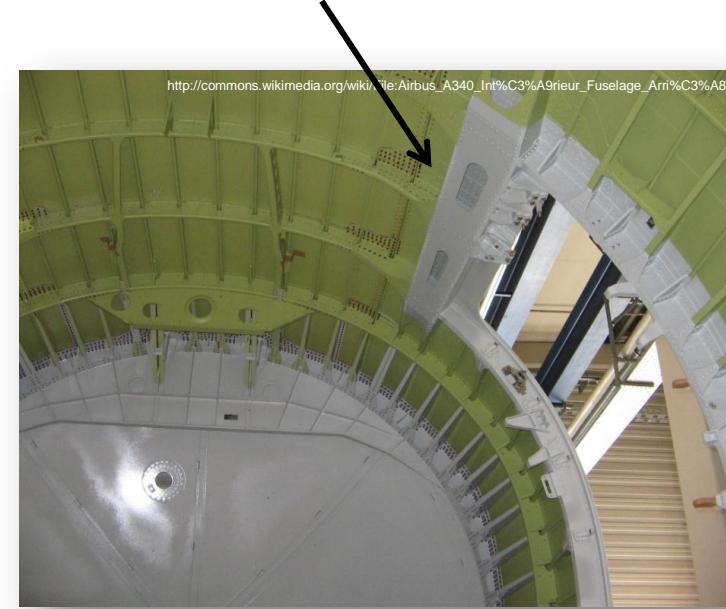
Fuselage structure

□ Cutout

The fuselage has requirements for opening cut-out such windows, doors, service panels, hatches, bomb bays...



The direct load paths are interrupted and as a result the structure around the cut-out must be reinforced to maintain the required strength



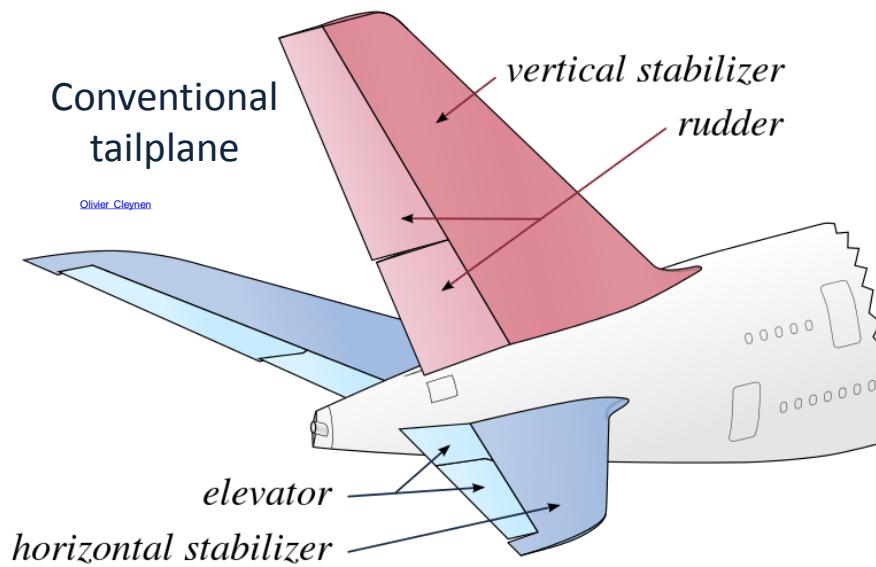


Stabilizers structure

□ Stabilizers



Alternate
tailplane: V-tail



Horizontal stabilizer of an Airbus A320

Stabilizer construction is similar to wing construction.

Stabilizers structure

□ Horizontal Stabilizers



Boeing B07

Box construction with spar, stronger ribs and skin with stringer

Low weight



Lockheed L-1011

□ Horizontal Stabilizers

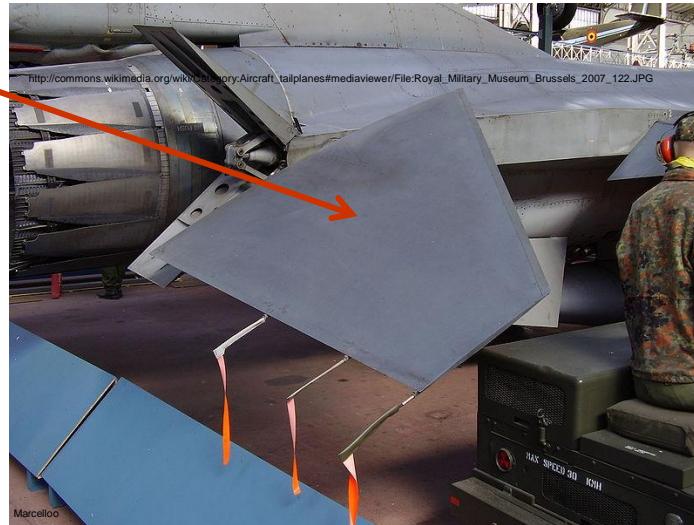
Two designs



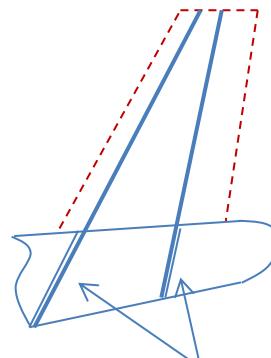
http://commons.wikimedia.org/wiki/File:F-16_Falcon_Fighting_-..._091119-F-7323C-292.jpg?uselang=es

Lockheed Martin F-16

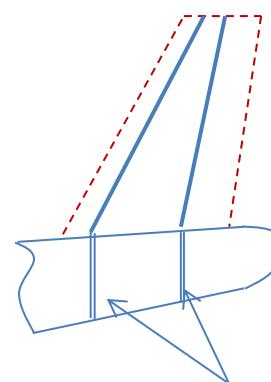
In fighter design,
usually no spar + ribs
construction is used



□ Vertical Stabilizers



Fuselage
bulkheads



Fuselage
bulkheads

Tail box front and rear
spar terminated at aft
fuselage bulkhead

Tail box front and
rear spar terminated
outside of fuselage

The span of the t-tail fin is aprox. 1/3 than
conventional tail

The vertical stabilizer must be made considerably
stronger and stiffer to support the forces generated
by the tailplane



http://commons.wikimedia.org/wiki/Category:Aircraft_tails#mediaviewer/File:Airforce_Museum_Berlin-Gatow_322.JPG



Russavia
[http://commons.wikimedia.org/wiki/File:N551VL_Gulfstream_G550_G-V-SP_\(9489975740\).jpg](http://commons.wikimedia.org/wiki/File:N551VL_Gulfstream_G550_G-V-SP_(9489975740).jpg)

- Aircraft Structures for engineering students**
H.G. Megson
Elsevier
Section B1 Principles of stressed skin construction
 Chapter 12 Structural components of aircraft
Section 12 B2 Airworthiness and airframe loads
 Chapter 13 Airworthiness
 Chapter 14 Airframe loads

- Airframe Structural Design**
M.C.Y. Niu
Hong Kong Commilit Press LTD
Section 3. Aircraft loads
 Chapter 8 Wing Box Structure
 Chapter 11 Fuselage