



Universidad  
Carlos III de Madrid  
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# Aerospace Structures

## Chapter 1. Structural description of the aircraft



## CHAPTER 1. Structural description of the aircraft

### Structural components of aircraft

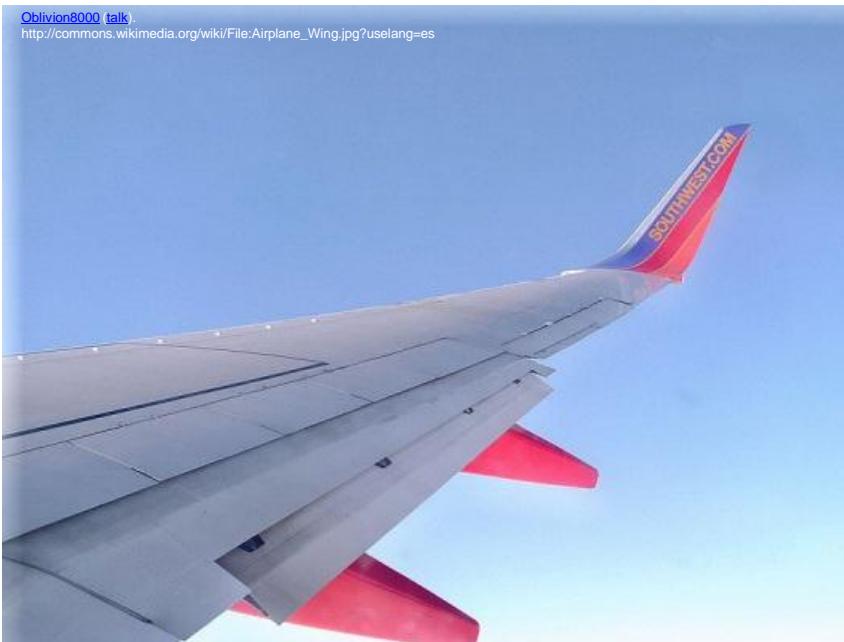
- Introduction
- Wing structure
- Fuselage structure
- Stabilizers structure
- References



## *Wing structure*

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## ■ Wing function



The function of the wing is to produce lift

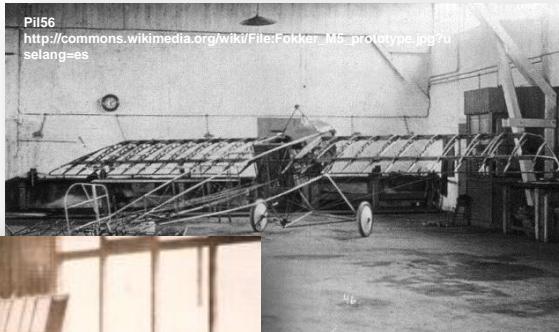
Main loads applied on the wing:

- Aerodynamic loads
- Weight

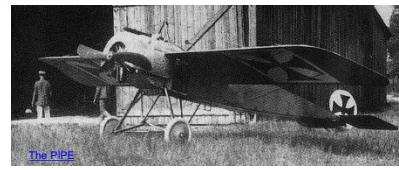
A wing works as a beam able to support the internal forces (Shear, bending and torsional moments)

## ■ Structures types

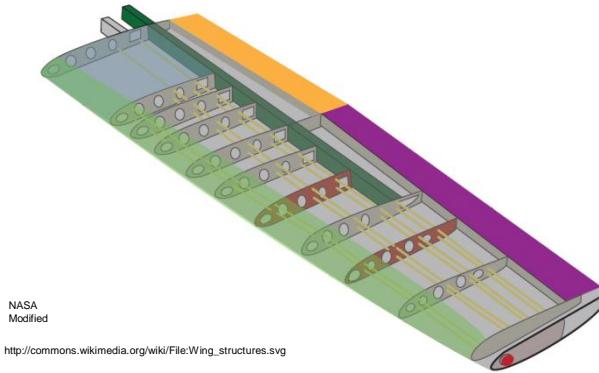
### Truss-type construction



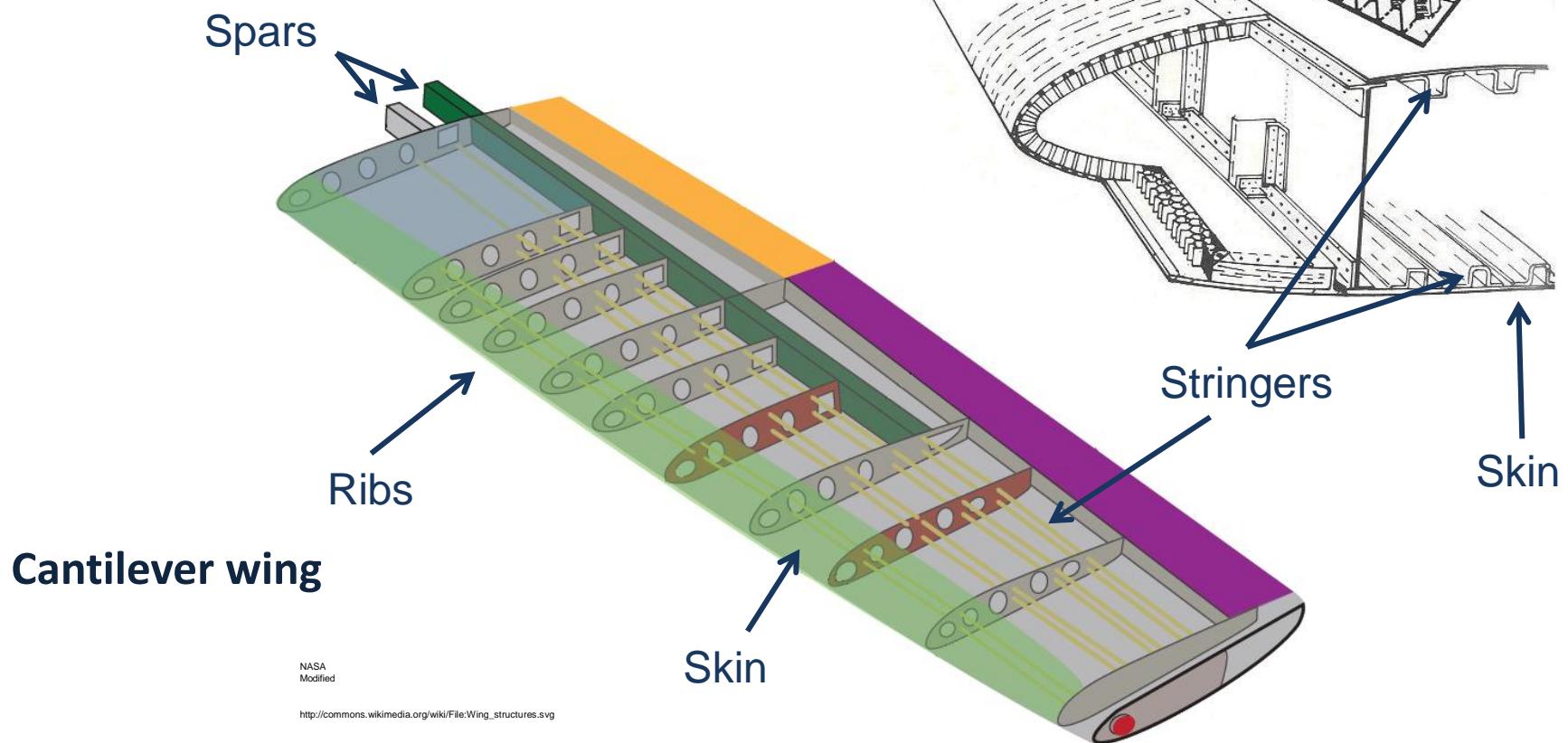
### Early wing structures



### Stressed-skin construction

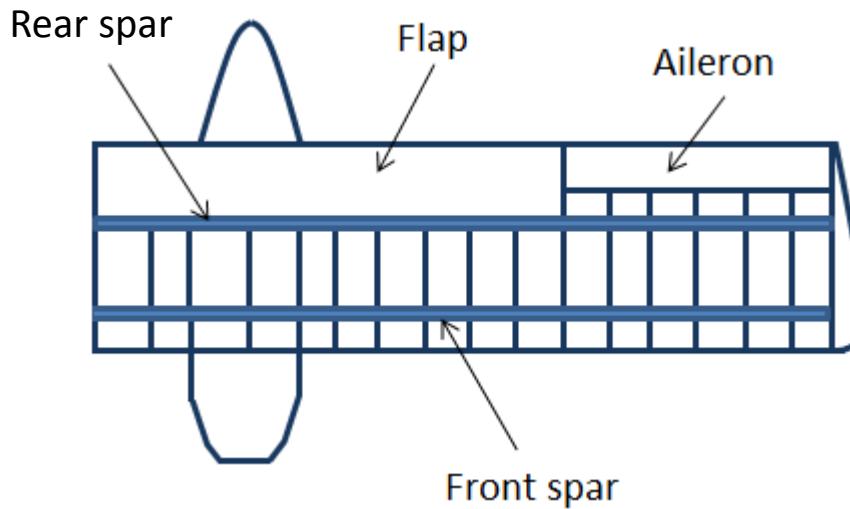


## ■ Wing elements



Principal structural elements of the wing

## ■ Spars



Torsion box structure: 2-3 spar

Front spar: 15-30% chord

Rear spar: 65-75% chord

## Semicantilever wings

Some aircraft have external struts for wing bracing



bus227  
<http://commons.wikimedia.org/wiki/File:FlyingWires.JPG>



CambridgeBayWeather



NASA/The Boeing Company  
[http://commons.wikimedia.org/wiki/File:Boeing\\_SUGAR\\_Volt\\_concept\\_aircraft\\_2010.jpg?uselang=es](http://commons.wikimedia.org/wiki/File:Boeing_SUGAR_Volt_concept_aircraft_2010.jpg?uselang=es)



Credit: NASA Langley/Sean Smith  
[http://www.nasa.gov/centers/langley/multimedia/iotw-tdt-wing\\_pr.htm](http://www.nasa.gov/centers/langley/multimedia/iotw-tdt-wing_pr.htm)

Boeing Aerodynamic Efficiency  
Improvement Joined Wing

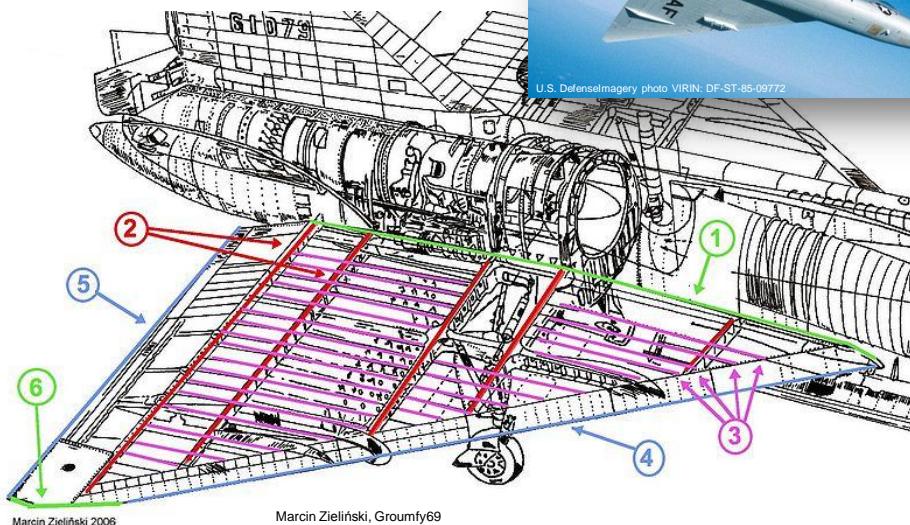


Boeing SUGAR Volt concept

# Wing structure

## ■ Spars

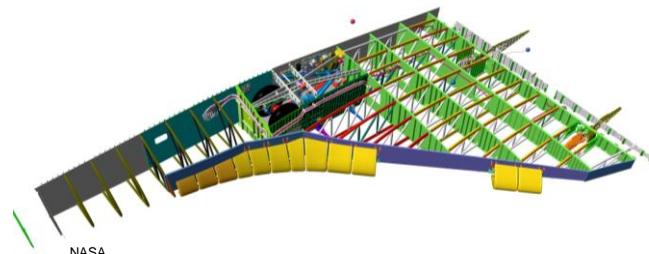
Convair F-106A Delta Dart



- (1) Wing root
- (2) Spars
- (3) Ribs
- (4) Leading edge
- (5) Trailing edge
- (6) Wing tip



Torsion box structure: multispar  
Parallel spars



NASA Photo ID: S81-30746  
[http://commons.wikimedia.org/wiki/File:Columbia\\_landing\\_on\\_Rogers\\_dry\\_lake.triddle.jpg](http://commons.wikimedia.org/wiki/File:Columbia_landing_on_Rogers_dry_lake.triddle.jpg)

## ■ Spars

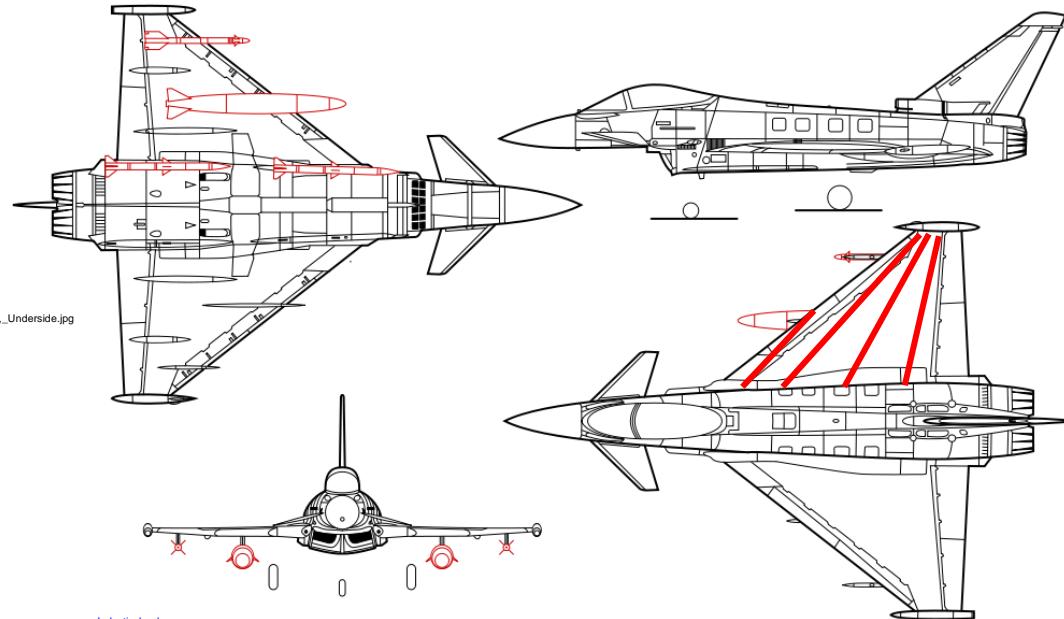


tMH

[http://commons.wikimedia.org/wiki/Category:Eurofighter\\_Typhoon?uselang=es#mediaviewer/File:Eurofighter\\_Typhoon,\\_Underside.jpg](http://commons.wikimedia.org/wiki/Category:Eurofighter_Typhoon?uselang=es#mediaviewer/File:Eurofighter_Typhoon,_Underside.jpg)

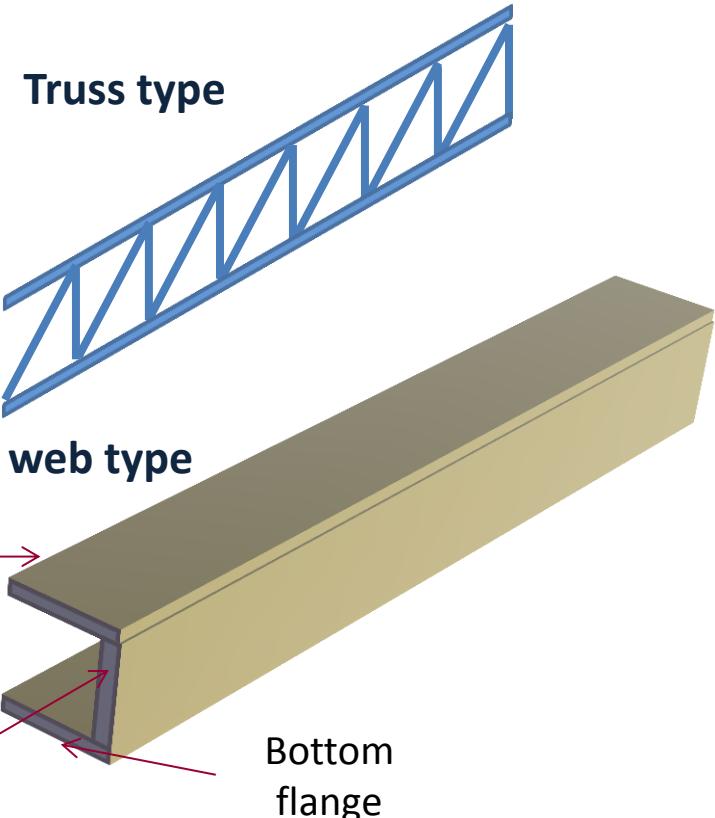
Eurofighter

Torsion box structure: multispar  
Converging spars

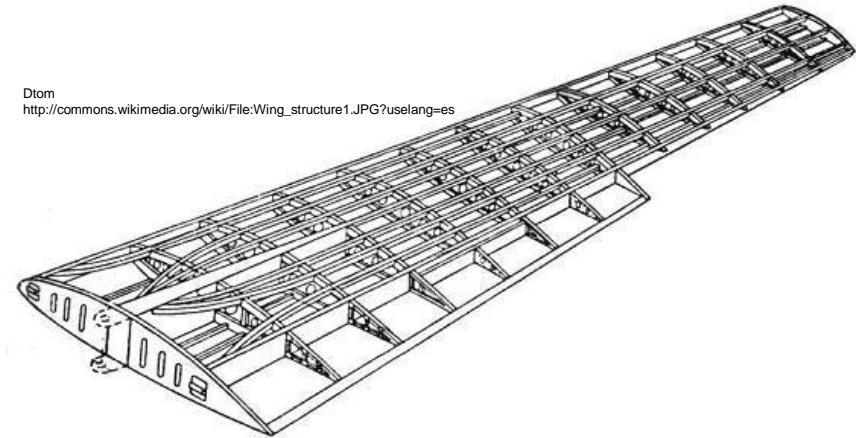


## ■ Spars

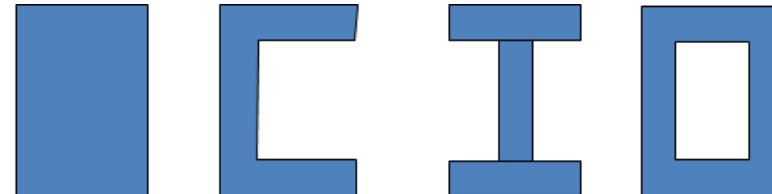
### Types of spar construction



Dtom  
[http://commons.wikimedia.org/wiki/File:Wing\\_structure1.JPG?uselang=es](http://commons.wikimedia.org/wiki/File:Wing_structure1.JPG?uselang=es)



### Cross-section geometries



Rectangular

Channel

I or double-T

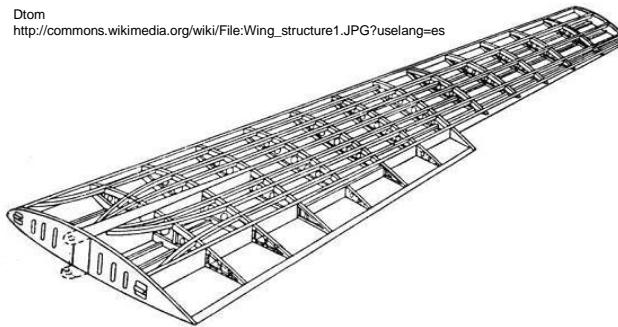
Box

# Wing structure

## ■ Ribs

Dtom

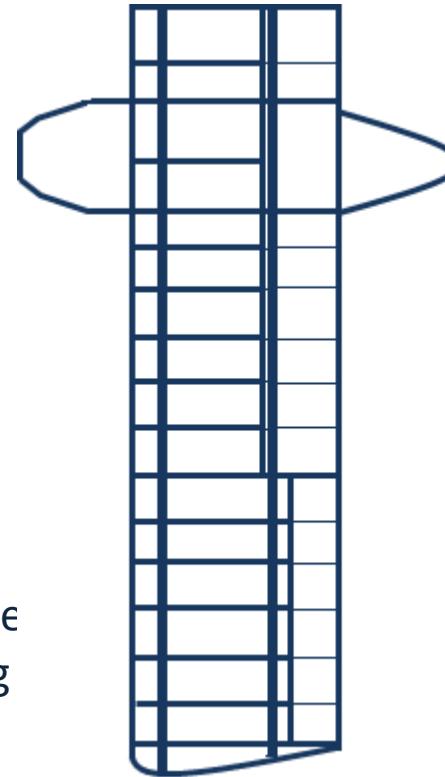
[http://commons.wikimedia.org/wiki/File:Wing\\_structure1.JPG?uselang=es](http://commons.wikimedia.org/wiki/File:Wing_structure1.JPG?uselang=es)



The wing ribs determine the shape and thickness of the wing

### Ribs types

- Shear web ribs
- Truss ribs



The rib spacing is selected after a optimization process (skin+stringer+ribs)

Rib are to be located at each aileron and flap hinge.

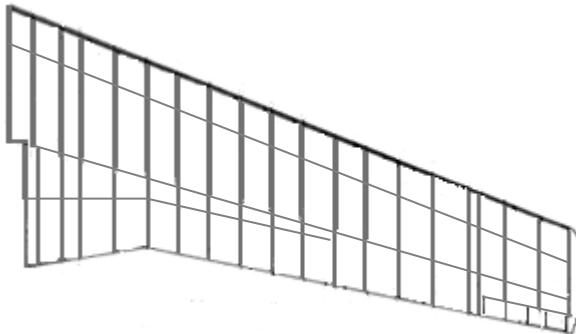
# Wing structure

## ■ Ribs arrangement



[http://commons.wikimedia.org/wiki/File:Aerospatiale\\_SE-210\\_Caravelle\\_10B3\\_Super\\_B,\\_Sterling\\_Airways\\_AN0018394.jpg](http://commons.wikimedia.org/wiki/File:Aerospatiale_SE-210_Caravelle_10B3_Super_B,_Sterling_Airways_AN0018394.jpg)?uselang=es

**Aerospatiale Caravelle**

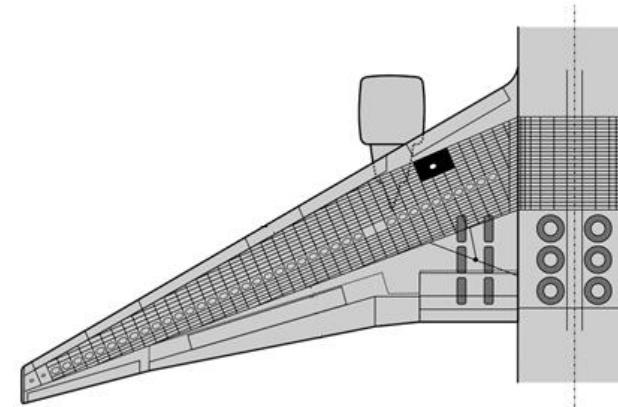


Wing ribs parallel to the flight path



[http://commons.wikimedia.org/wiki/File:Boeing\\_767#mediaviewer/File:LOT\\_Polish\\_Airlines,\\_Boeing\\_767-25D\\_ER,,Mississauga\\_\(295649204\).jpg](http://commons.wikimedia.org/wiki/File:Boeing_767#mediaviewer/File:LOT_Polish_Airlines,_Boeing_767-25D_ER,,Mississauga_(295649204).jpg)

**Boeing 767**

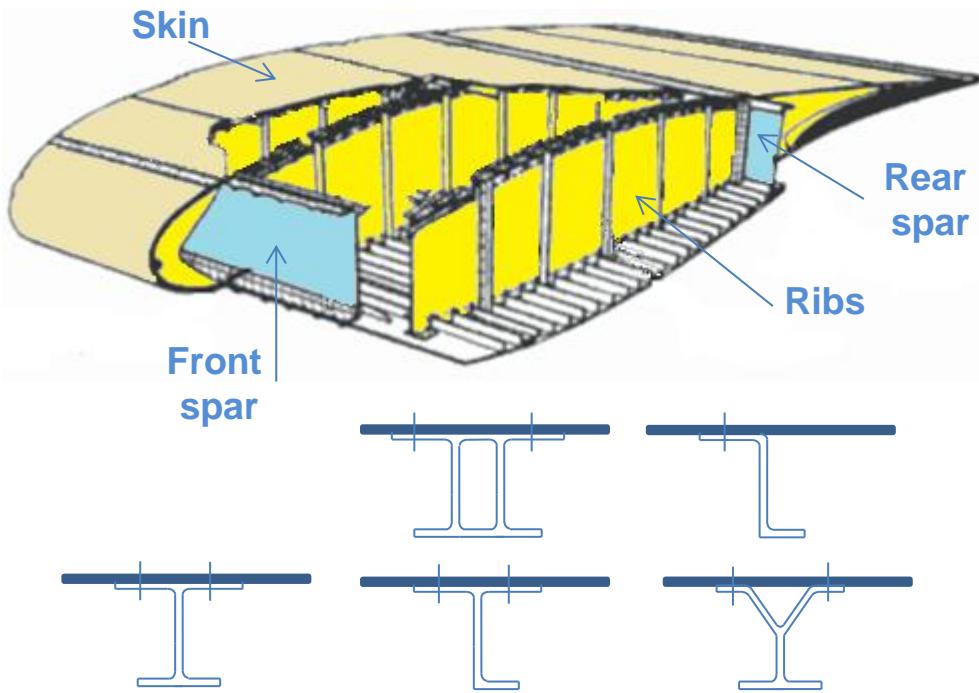


Tosaka  
[http://commons.wikimedia.org/wiki/File:Jetliner%27s\\_wing\\_structure\\_\(B-777\).PNG](http://commons.wikimedia.org/wiki/File:Jetliner%27s_wing_structure_(B-777).PNG)

Rib perpendicular to the spars

## ■ Skin

50-70% of the structural weight of the wing



Typical wing skin-stringer panels

It transmits aerodynamic forces to the longitudinal and transverse members

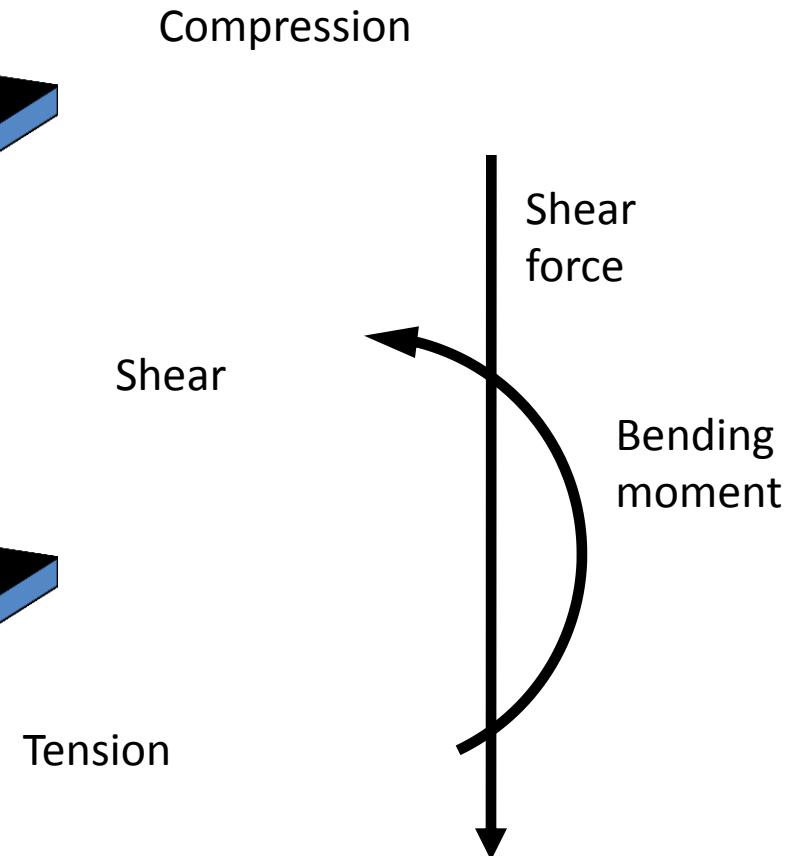
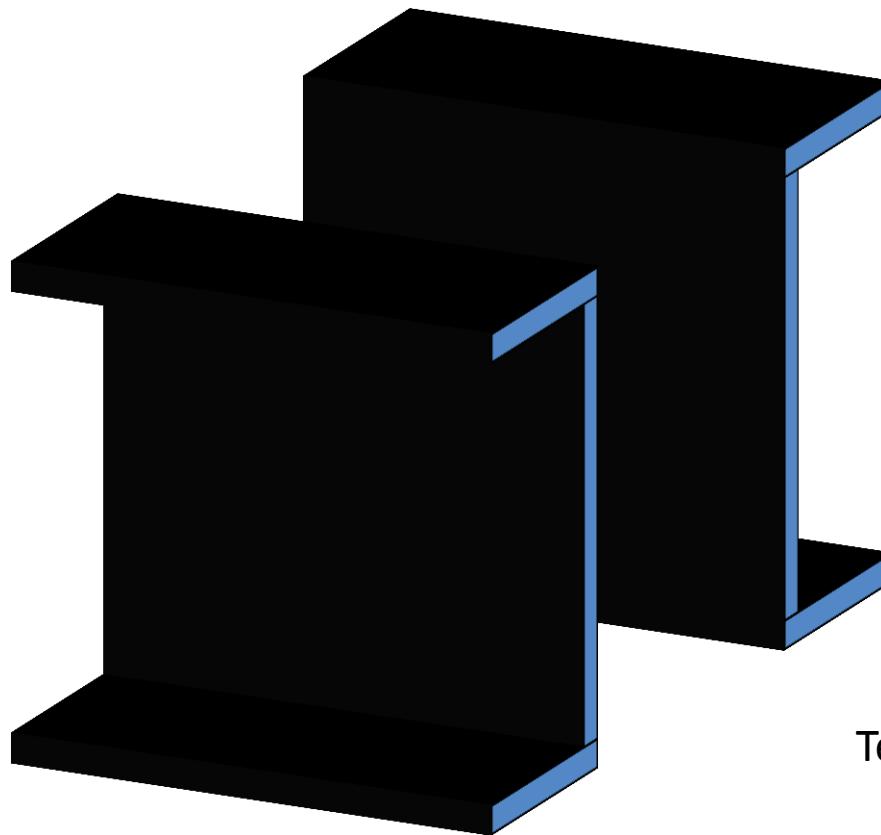
In some designs the spar caps resist the bending stresses, in other skin is the bending-load resistant element

## Integrally stiffened panels

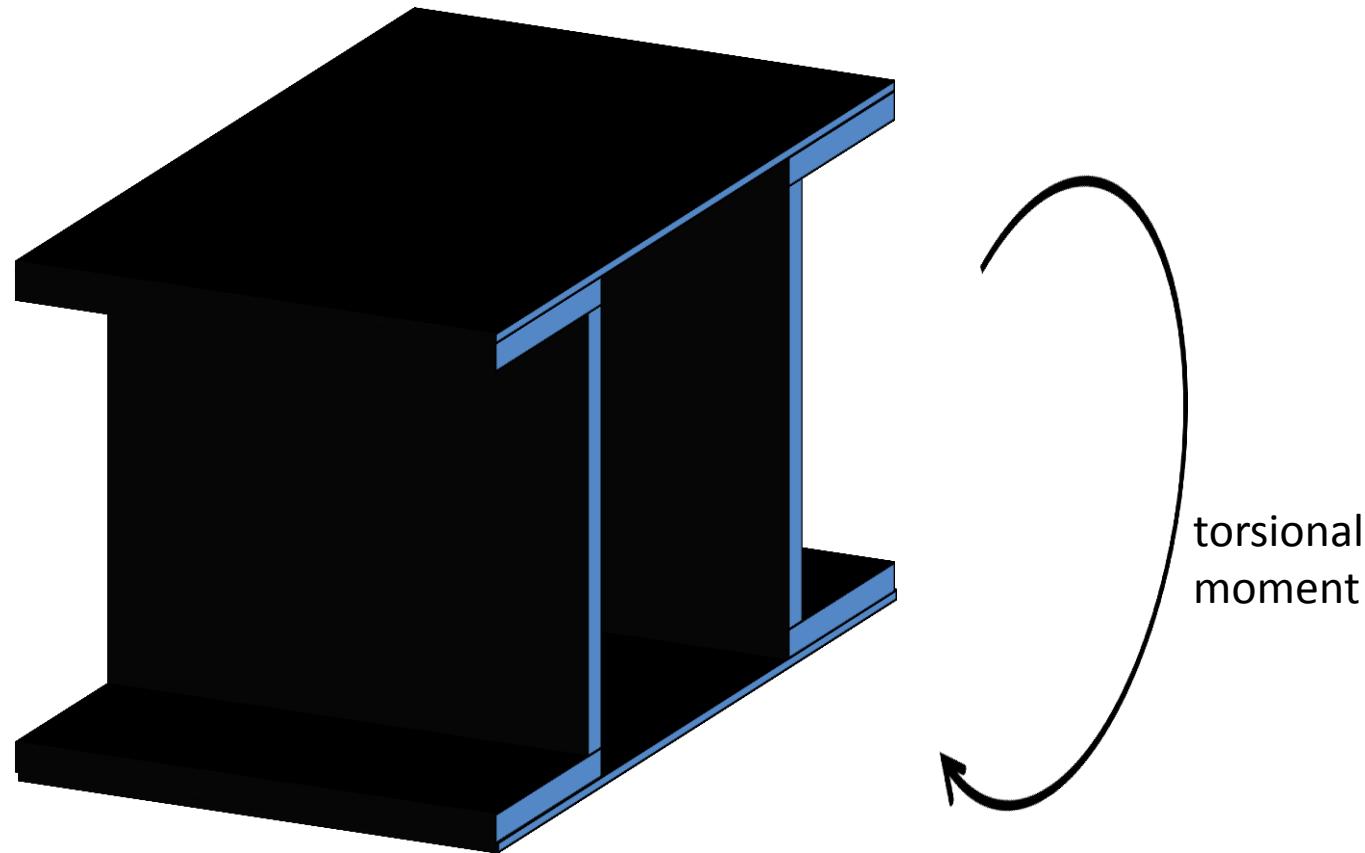


Wing skins are machined from a thick plate to obtain the skin-stringer geometry

## ■ Wing box structure



## ■ Wing box structure





## *Fuselage structure*

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# Fuselage structure

## ■ Fuselage functions



The fuselage function is to carry the payload and support many the aircraft system. They connect the other structural element (wing and empennage)



### Loads on the fuselage:

- Wing reactions
- Landing gear reactions
- Internal pressure

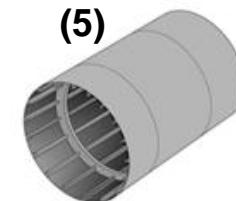
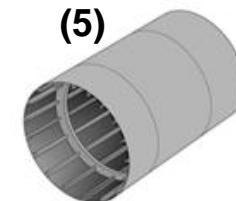
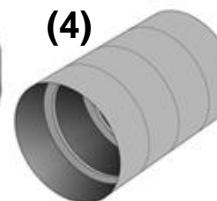
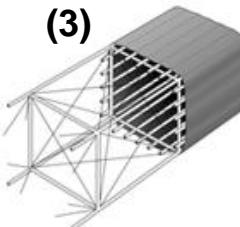
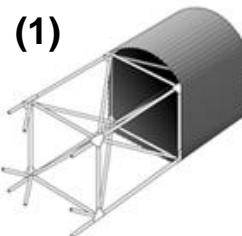
## ■ Fuselage types

### Non stressed-skin structures

- Truss fuselage (1)
- Geodetic fuselage (2)

### Stressed-skin Structures

- Corrugated (3)
- Monocoque fuselage (4)
- Semi-Monocoque fuselage (5)



Author: Tosaka  
[http://commons.wikimedia.org/wiki/File:Airframe\\_\(4\\_types\).PNG?uselang=es](http://commons.wikimedia.org/wiki/File:Airframe_(4_types).PNG?uselang=es)

**Blériot XI**



[http://commons.wikimedia.org/wiki/B%CC%A9ri%C3%A9ot\\_XI?uselang=es#mediaviewer/File:Bleriot\\_XI\\_Thulin\\_2.jpg](http://commons.wikimedia.org/wiki/B%CC%A9ri%C3%A9ot_XI?uselang=es#mediaviewer/File:Bleriot_XI_Thulin_2.jpg)

**Vickers Wellington**



[http://commons.wikimedia.org/wiki/File:Vickers\\_Wellington\\_Mk2.jpg](http://commons.wikimedia.org/wiki/File:Vickers_Wellington_Mk2.jpg)

**Ford Trimotor**



[http://commons.wikimedia.org/wiki/File:Ford\\_Trimotor\\_EAA.jpg?uselang=es](http://commons.wikimedia.org/wiki/File:Ford_Trimotor_EAA.jpg?uselang=es)

**Rutan VariEze**



[http://commons.wikimedia.org/wiki/Category:Rutan\\_VariEze#mediaviewer/File:VariEze\\_in\\_flight.jpg](http://commons.wikimedia.org/wiki/Category:Rutan_VariEze#mediaviewer/File:VariEze_in_flight.jpg)

**Airbus A350**



[http://commons.wikimedia.org/wiki/Airbus\\_A350#mediaviewer/File:A350\\_First\\_Flight\\_-\\_Low\\_pass\\_03.jpg](http://commons.wikimedia.org/wiki/Airbus_A350#mediaviewer/File:A350_First_Flight_-_Low_pass_03.jpg)

# Fuselage structure

## ■ Fuselage types

### Truss fuselage



Philip Capon

[http://commons.wikimedia.org/wiki/File:Sopwith\\_Camel\\_taking\\_off,\\_Masterton,\\_New\\_Zealand,\\_April\\_2009.jpg?uselang=es](http://commons.wikimedia.org/wiki/File:Sopwith_Camel_taking_off,_Masterton,_New_Zealand,_April_2009.jpg?uselang=es)

### Sopwith Camel, 1917

Most of the early aircraft used this kind of construction



The diagonal elements can be bars or wires

## □ Fuselage types

### Truss fuselage

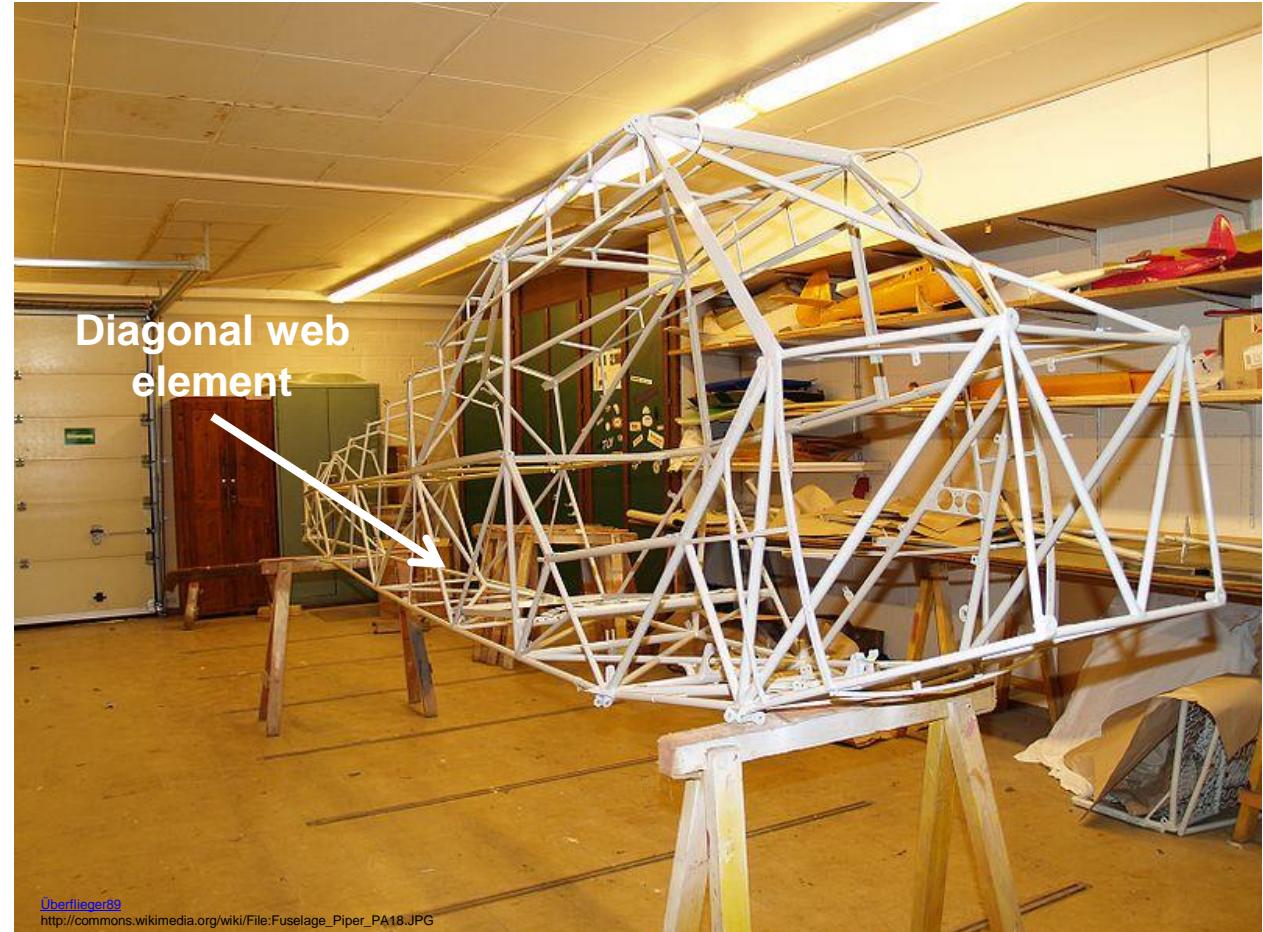


[http://commons.wikimedia.org/wiki/Category:Super\\_18\\_Model\\_S18#mediaviewer/File:Super18-Super18-180.jpg](http://commons.wikimedia.org/wiki/Category:Super_18_Model_S18#mediaviewer/File:Super18-Super18-180.jpg)

Photo: FlugKerl2

Piper PA-18

This type of structures are still used in small aircraft



## □ Fuselage types

### Truss fuselage

Non structural skin



[http://commons.wikimedia.org/wiki/Piper\\_PA-18#mediaviewer/File:L18FuselageCUT.jpg](http://commons.wikimedia.org/wiki/Piper_PA-18#mediaviewer/File:L18FuselageCUT.jpg)

Überflieger89



[http://commons.wikimedia.org/wiki/Piper\\_PA-18#mediaviewer/File:L18FuselageCT.jpg](http://commons.wikimedia.org/wiki/Piper_PA-18#mediaviewer/File:L18FuselageCT.jpg)

Überflieger89

# Fuselage structure

## □ Fuselage types

### Geodetic fuselage



Vickers Wellington  
1936



Cobaltor  
[http://commons.wikimedia.org/wiki/File:Vickers\\_Wellington\\_Mark\\_X,\\_HE239\\_%27NA-Y%27,\\_of\\_No.\\_428\\_Squadron,\\_RCAF\\_\(April\\_1943\).png?uselang=es](http://commons.wikimedia.org/wiki/File:Vickers_Wellington_Mark_X,_HE239_%27NA-Y%27,_of_No._428_Squadron,_RCAF_(April_1943).png?uselang=es)

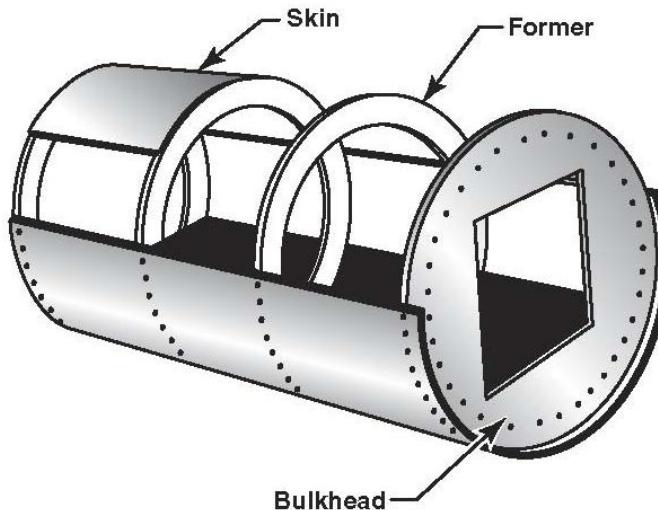


In this type of construction multiple flat strip stringers are wound about the formers in opposite spiral directions

# Fuselage structure

## □ Fuselage types

Eas4200c.f08.nine.o  
[http://commons.wikimedia.org/wiki/File:Chapter\\_1\\_img\\_26.jpg?uselang=es](http://commons.wikimedia.org/wiki/File:Chapter_1_img_26.jpg?uselang=es)



The skin supports the primary stresses



Deperdussin monoplane racer  
1912

First monocoque design

Designed by Louis Béchereau

### Monocoque fuselage

Virtually no internal framework

Only transversal elements (former) + skin

Stoddard-Hamilton Glasair III



[http://commons.wikimedia.org/wiki/Category:Stoddard-Hamilton\\_Glasair\\_III#mediaviewer/File:GlasairIII-Landing.jpg](http://commons.wikimedia.org/wiki/Category:Stoddard-Hamilton_Glasair_III#mediaviewer/File:GlasairIII-Landing.jpg)

Used today in small aircrafts with complete composite structure

# Fuselage structure

## □ Fuselage types

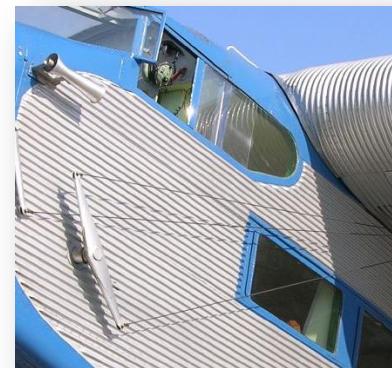


[http://commons.wikimedia.org/wiki/File:Ju-Air\\_Junkers\\_Ju-52-3m\\_HB-HOS\\_Hahnweide\\_2011\\_03.jpg?uselang=es](http://commons.wikimedia.org/wiki/File:Ju-Air_Junkers_Ju-52-3m_HB-HOS_Hahnweide_2011_03.jpg?uselang=es)

Junkers Ju-52/3m  
1932



[http://commons.wikimedia.org/wiki/File:Ju52\\_civil.jpg?uselang=es](http://commons.wikimedia.org/wiki/File:Ju52_civil.jpg?uselang=es)



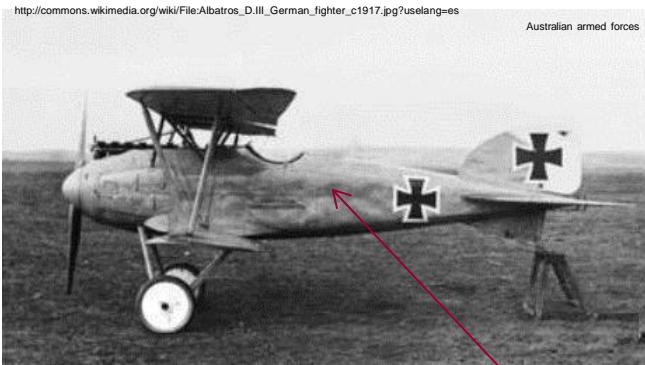
Ford Tri-Motor  
1926



[http://commons.wikimedia.org/wiki/File:Ford\\_Tri-Motor\\_\(EAA\)\\_%28340406538%29.jpg?uselang=es](http://commons.wikimedia.org/wiki/File:Ford_Tri-Motor_(EAA)_%28340406538%29.jpg?uselang=es)

It had a structure with  
metallic corrugated skin

## □ Fuselage types



Albatros D.III  
1916

Streamlined fuselage

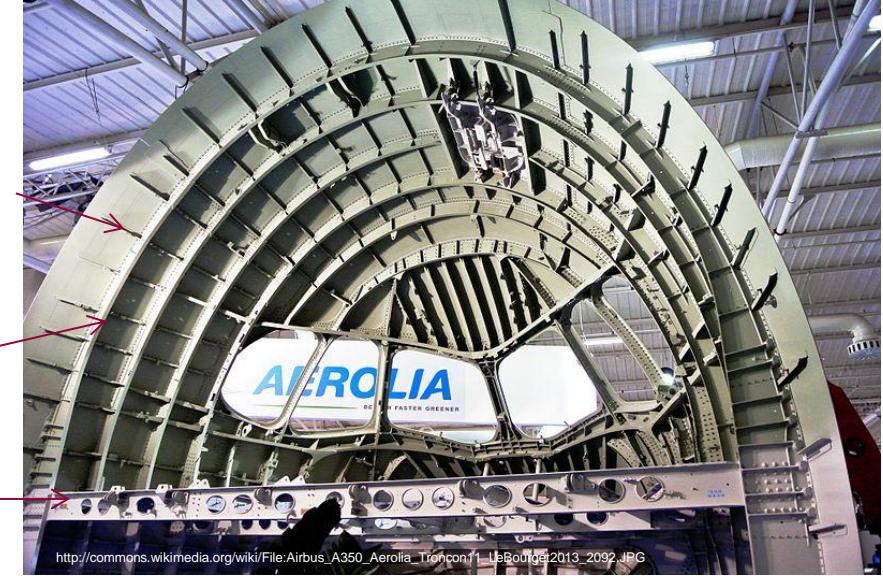
### Longitudinal elements

- Longerons
- Stringers

### Transverse element

- Frames
- Bulkheads

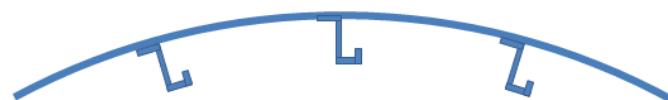
### Semi-Monocoque fuselage



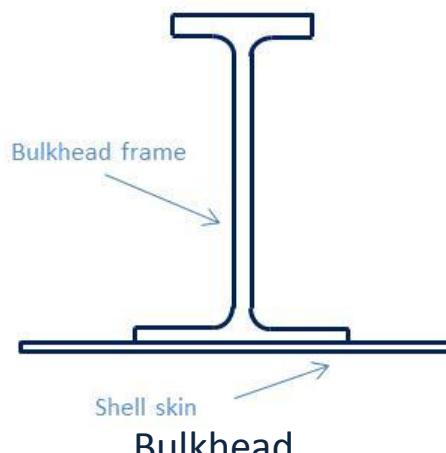
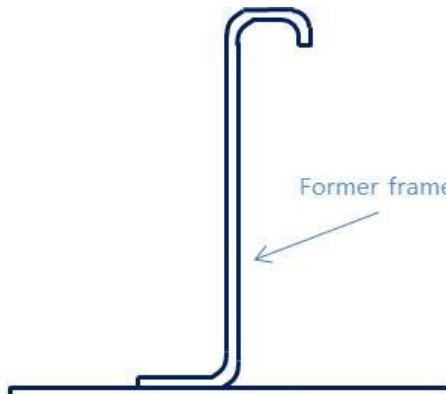
## □ Skin and stringer

The skin carries the shear stresses produced by the internal forces (torque and shear force)

**Typical fuselage  
skin-stringer panels**



## □ Frames



Author: Kolossos  
<http://commons.wikimedia.org/wiki/File:Fuselage-747.jpg?uselang=es>



Distance between frames are around 500 mm

Frame thickness are 75-150 mm

### Frames function

- Maintain the shape of the fuselage
- Increase the buckling strength of the stringer



Bibi95  
[http://commons.wikimedia.org/wiki/File:Exposition\\_-\\_Les\\_100\\_ans\\_de\\_l%C2%A9rospace\\_-\\_Paris\\_-\\_4\\_Octobre\\_2008\\_\(2913799759\).jpg](http://commons.wikimedia.org/wiki/File:Exposition_-_Les_100_ans_de_l%C2%A9rospace_-_Paris_-_4_Octobre_2008_(2913799759).jpg)

## □ Frames

Bibi95

[http://upload.wikimedia.org/wikipedia/commons/1/19/Exposition\\_Les\\_100\\_ans\\_de\\_l\\_aerospatiale.\\_-Paris.\\_-4\\_Octobre\\_2008\\_%282913796295%29.jpg](http://upload.wikimedia.org/wikipedia/commons/1/19/Exposition_Les_100_ans_de_l_aerospatiale._-Paris._-4_Octobre_2008_%282913796295%29.jpg)



Front dome

Rear dome



Frames →

### Types of frames

- Former frame
- Bulkhead
- domes

NASA  
[http://commons.wikimedia.org/wiki/File:Shuttle\\_Carrier\\_Aircraft\\_interior\\_bulkhead.jpg](http://commons.wikimedia.org/wiki/File:Shuttle_Carrier_Aircraft_interior_bulkhead.jpg)

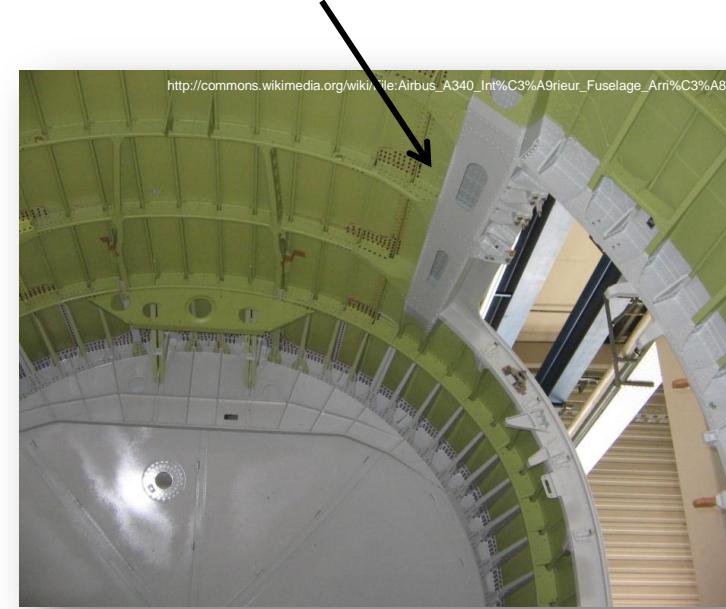
# Fuselage structure

## □ Cutout

The fuselage has requirements for opening cut-out such windows, doors, service panels, hatches, bomb bays...



The direct load paths are interrupted and as a result the structure around the cut-out must be reinforced to maintain the required strength





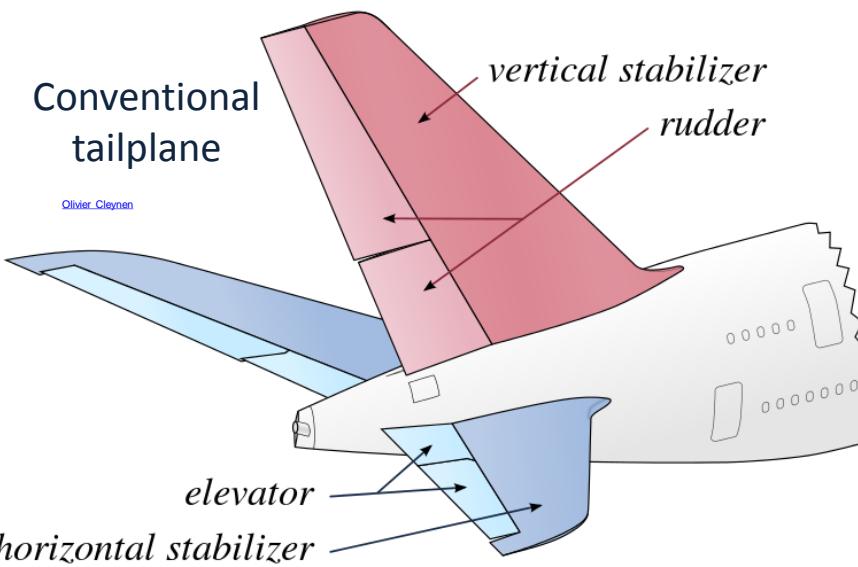
## *Stabilizers structure*

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## □ Stabilizers



Alternate  
tailplane: V-tail



Horizontal stabilizer of an Airbus A320

Stabilizer construction is similar to wing construction.

# Stabilizers structure

## □ Horizontal Stabilizers



### Boeing B07

Box construction with spar, stronger ribs and skin with stringer

Low weight



### Lockheed L-1011

## □ Horizontal Stabilizers

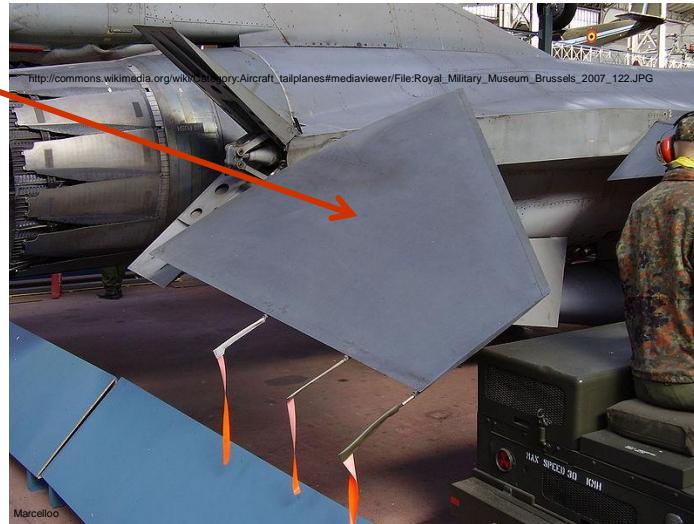
Two designs



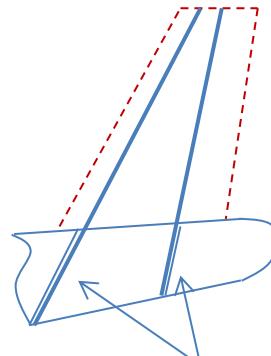
[http://commons.wikimedia.org/wiki/File:F-16\\_Falcon\\_Fighting\\_-\\_091119-F-7323C-292.jpg?uselang=es](http://commons.wikimedia.org/wiki/File:F-16_Falcon_Fighting_-_091119-F-7323C-292.jpg?uselang=es)

### Lockheed Martin F-16

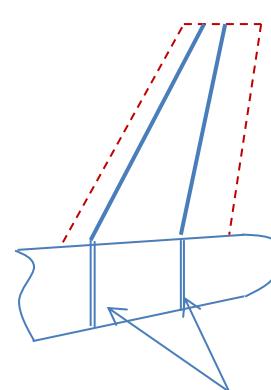
In fighter design,  
usually no spar + ribs  
construction is used



## □ Vertical Stabilizers



Fuselage  
bulkheads



Fuselage  
bulkheads

Tail box front and rear  
spar terminated at aft  
fuselage bulkhead

Tail box front and  
rear spar terminated  
outside of fuselage

The span of the t-tail fin is aprox. 1/3 than  
conventional tail

The vertical stabilizer must be made considerably  
stronger and stiffer to support the forces generated  
by the tailplane



[http://commons.wikimedia.org/wiki/Category:Aircraft\\_tails#mediaviewer/File:Airforce\\_Museum\\_Berlin-Gatow\\_322.JPG](http://commons.wikimedia.org/wiki/Category:Aircraft_tails#mediaviewer/File:Airforce_Museum_Berlin-Gatow_322.JPG)



Russavia  
[http://commons.wikimedia.org/wiki/File:N551VL\\_Gulfstream\\_G550\\_G-V-SP\\_\(9489975740\).jpg](http://commons.wikimedia.org/wiki/File:N551VL_Gulfstream_G550_G-V-SP_(9489975740).jpg)

- Aircraft Structures for engineering students**  
H.G. Megson  
Elsevier  
Section B1 Principles of stressed skin construction  
    Chapter 12 Structural components of aircraft  
Section 12 B2 Airworthiness and airframe loads  
    Chapter 13 Airworthiness  
    Chapter 14 Airframe loads
  
- Airframe Structural Design**  
M.C.Y. Niu  
Hong Kong Commilit Press LTD  
Section 3. Aircraft loads  
    Chapter 8 Wing Box Structure  
    Chapter 11 Fuselage